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Highlights

- Up-to-date reported test and numerical results of chord sidewall failure in RHS joints were collated.
- Effects of brace-to-chord height ratio, brace angle, steel grade and chord stress ratio were evaluated.
- Two design methods were proposed for chord sidewall failure in RHS joints under brace axial compression.
- Design of chord sidewall failure in RHS joints under brace axial tension and brace bending was discussed.

Design of chord sidewall failure in RHS joints using steel grades up to S960

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Abstract: It is well known that the current design rules adopted by international design codes such as ISO 14346 and design guides, e.g., the CIDECT design guide No. 3, for chord sidewall failure in mild steel RHS joints under brace axial compression are considerably conservative, if the RHS joints are adequately supported out-of-plane. This paper presents an investigation into chord sidewall failure in rectangular hollow section (RHS) joints using steel grades up to S960. Representative existing design methods for chord sidewall failure in RHS joints are reviewed, and two alternative design methods, i.e., the modified bearing-buckling method and the Lan-Kuhn method, are proposed. Up-to-date test and numerical results reported in the literature are compiled. A wide range of geometric parameters, steel grades up to S960 and loading cases of brace axial loading, brace in-plane bending and brace out-of-plane bending are covered. The existing and proposed design methods are assessed against the collated results. The effects of brace-to-chord height ratio, brace angle, steel grade and chord stress ratio are evaluated. It is shown that the proposed design methods can provide more consistent resistance predictions for chord sidewall failure in mild steel and high-strength steel RHS joints under brace axial compression. Corresponding user-friendly design rules are suggested. The design of chord sidewall failure in RHS joints under brace axial tension, brace in-plane bending and brace out-of-plane bending is discussed. Further required research on, in particular, high-strength steel RHS joints is highlighted.

Keywords: Rectangular hollow section; X joints; T joints; Y joints; Chord sidewall failure; Design rules

1. Introduction

Rectangular hollow sections (RHS) exhibit an aesthetic appearance and feature excellent structural efficiency especially with regard to loading of compression and torsion because of the closed shape. The evident advantages of RHS result in wide applications in structural, mechanical, transport and offshore fields. The connection of RHS members is vitally crucial for the structural integrity, and direct welding of the intersecting brace to the through chord is the simplest and cleanest solution for the connection. Design rules are needed for such welded RHS joints to facilitate structural applications.

Fig. 1 shows the configurations and notations of RHS-to-RHS X, T and Y joints. Chord sidewall failure is a typical failure mode in full-width RHS joints with brace-to-chord width ratio (β) of 1.0. In the 1970s, test data for chord sidewall failure in RHS X joints became available from Czechowski and Brodka [1] and Barentse [2]. Czechowski

41 and Brodka [1] developed an empirical equation based on their data which showed a large scatter. This is probably
42 because in the Polish tests the brace and chord of the X joints were fabricated from cold-formed channel sections
43 with fabrication tolerances. Furthermore, distortion of the chord cross-section resulted in a sway-type failure mode
44 because of the pinned-end support at the brace ends and the inadequate out-of-plane support at the chord ends.
45 Barentse [2] assessed various local buckling models against his test results. Brodka and Szlendak [3] and Kato and
46 Nishiyama [4] proposed analytical models which appear to be too complicated for design. Later, Wardenier and
47 Davies [5] developed a simpler combined bearing-buckling model based on a conservative lower bound of the
48 aforementioned Polish and Dutch test results. It adopts a combined check for the bearing resistance using the steel
49 yield stress of the chord (f_{y0}) and the local buckling capacity employing a local buckling stress of the chord sidewall
50 (f_k). The value of f_k can be determined using the relevant Eurocode buckling curves [6] or equivalent buckling
51 curves. The bearing-buckling method is adopted by various design codes, e.g., EN 1993-1-8 [7] and ISO 14346
52 [8], and design guides such as the CIDECT design guide No. 3 [9-10] and the IIW recommendations [11-13] for
53 chord sidewall failure.

54
55 Extensive research on chord sidewall failure has been conducted since the mid-1980s. Davies et al. [14]
56 summarised various design methods and the influence of different joint parameters for chord sidewall failure in
57 RHS X joints. Packer [15] conducted tests on 31 full-width RHS X joints to supplement the existing test database
58 of 40 RHS X joints reported in the literature, and concluded that the codified bearing-buckling method for chord
59 sidewall failure was too conservative. Giddings and Wardenier [16] compiled CIDECT Monograph No. 6 in which
60 various state-of-the-art theories at that time for chord sidewall failure were summarised. Davies and Roodbaraky
61 [17] examined the effect of brace angle (θ_1) on the resistances of various failure modes in RHS X joints using the
62 results of tests as well as elastic and elastic-plastic numerical analyses reported by Platt [18]. It was found that the
63 average resistance enhancement for decreasing brace angles could be quantified by the brace angle function of
64 $(1/\sin\theta_1)^{0.5}$ for chord sidewall failure in RHS X joints under brace axial compression and tension. Yu [19] proposed
65 a four-hinge yield line model and assumed that the chord sidewall was fully clamped for chord sidewall failure in
66 RHS-to-RHS X and T joints subjected to brace axial compression, brace in-plane bending and brace out-of-plane
67 bending. Becque and Cheng [20] conservatively assumed that the chord sidewall is pinned along the chord length
68 direction, and proposed a plate buckling model to predict the buckling initiation of the chord sidewall in RHS-to-
69 RHS X joints. Kuhn et al. [21] proposed an equation of the buckling reduction factor which is linearized against
70 chord height to wall thickness ratio ($2\gamma^*=h_0/t_0$) for chord sidewall failure in RHS X joints under brace axial
71 compression. This simplifies the determination of f_k values without using the buckling curves. Wardenier [22]
72 proposed to modify the codified resistance equation to consider the effect of brace-to-chord height ratio ($\eta^*=h_1/h_0$).
73 Lan et al. [23-24] developed an analytical model for plate buckling to properly consider the beneficial restraint of
74 the chord face and brace for the chord sidewall in RHS-to-RHS X and T joints. Comprehensive assessment of the
75 design methods remains limited, and more suitable design rules for chord sidewall failure in RHS joints are needed.

76
77 This study aims to evaluate existing design methods and to propose suitable design methods and design rules for
78 chord sidewall failure in mild steel and high-strength steel RHS joints. Test and numerical results of RHS X and
79 T joints reported in the literature have been collated. A wide range of geometric parameters, steel grades up to
80 S960 and loading cases of brace axial loading, brace in-plane bending and brace out-of-plane bending were

81 covered. Existing design methods and proposed design methods in this study were assessed against the compiled
82 results. The effects of brace-to-chord height ratio, brace angle, steel grade and chord stress were evaluated. Design
83 rules were proposed for chord sidewall failure in mild steel and high-strength steel RHS joints under brace axial
84 compression. The design of chord sidewall failure in RHS joints under brace axial tension, brace in-plane bending
85 and brace out-of-plane bending was discussed. Further research on, in particular, high-strength steel RHS joints
86 was highlighted.

87

88 **2. Design methods for chord sidewall failure**

89

90 **2.1. General**

91

92 This section elaborates the representative design methods in the literature and the proposed design methods in this
93 study for chord sidewall failure in RHS joints. The bearing-buckling model proposed by Wardenier and Davies [5]
94 is widely adopted by international design codes, e.g., EN 1993-1-8 [7] and ISO 14346 [8], and design guides such
95 as the CIDECT design guide No. 3 [9-10] and the IIW recommendations [11-13]. The design rules specified in
96 these design codes and design guides are nearly the same for chord sidewall failure in mild steel RHS joints. Kuhn
97 et al. [21] and Wardenier [22], among others, proposed modifications to the codified design method in order to
98 reduce the conservatism and scatter of the resistance predictions. Other analytical models were also proposed for
99 chord sidewall failure, e.g., the four-hinge yield line model in combination with a reduced chord sidewall buckling
100 length proposed by Yu [19], and the plate buckling models proposed by Becque and Cheng [20] and Lan et al.
101 [23-24]. Two alternative design methods, i.e., a modified bearing-buckling method and a so-called “Lan-Kuhn
102 method” are proposed herein. These design proposals are summarised in the subsequent sections. It is noted that
103 steel with a grade up to S355 is defined as mild steel in this study.

104

105 **2.2. Codified bearing-buckling model**

106

107 Fig. 2 shows the codified bearing-buckling model for chord sidewall failure in RHS joints under brace axial loading
108 [7, 8, 25]. It is based on a combined check for the bearing resistance using the steel yield stress of the chord (f_{y0})
109 and the local buckling capacity employing the local buckling stress of the chord sidewall (f_k). The f_k values can be
110 obtained using the relevant Eurocode buckling curves [6] or equivalent buckling curves. Chord sidewall failure is
111 conservatively considered as the buckling of a pinned-end strut with a buckling length of h_0-2t_0 . The spreading of
112 the normal component of the brace load ($N_1 \sin \theta_1$) is assumed to be over a length of $h_1/\sin \theta_1 + 5t_0$ at each chord
113 sidewall with a dispersion slope of 2.5 to 1 through the chord thickness. This results in the following basic
114 resistance equation for chord sidewall failure in mild steel RHS joints under brace axial loading:

$$115 N_{1,Rd} = \frac{f_k t_0}{\sin \theta_1} \left(\frac{2h_1}{\sin \theta_1} + 10t_0 \right) Q_f \quad (1)$$

116 where t_0 is the chord sidewall thickness, h_1 is the brace height, θ_1 is the brace angle (see Fig. 2) and Q_f is a chord
117 stress function which accounts for the effect of longitudinal chord stresses. The term f_k , which equals f_{y0} for brace
axial tension, is the buckling stress of the chord sidewall for brace axial compression, and is taken as [7, 8, 25]:

$$f_k = \begin{cases} 0.8\chi_C f_{y0} \sin \theta_1 & \text{for X joints} \\ \chi_C f_{y0} & \text{for T and Y joints} \end{cases} \quad (2)$$

118 where χ_C is a buckling reduction factor for column buckling according to EN 1993-1-1 [6], or a comparable design
119 code, for a normalized slenderness (λ_C) defined by [7, 8, 25]:

$$\lambda_C = \frac{3.46 \left(\frac{h_0}{t_0} - 2 \right) \sqrt{\frac{1}{\sin \theta_1}}}{\pi \sqrt{\frac{E}{f_{y0}}}} \quad (3)$$

120 The Eurocode buckling reduction factor (χ_C) can be obtained from tables as a function of the normalized
121 slenderness or by substituting Eq. (3) into Eqs. (4-5) where α is an imperfection factor. For cold-formed steel cross-
122 sections, a buckling curve c with $\alpha=0.49$ is used, and a buckling curve a with $\alpha=0.21$ is adopted for hot-finished
123 steel cross-sections using steel grades up to S420.

$$\chi = \frac{1}{\varphi + \sqrt{\varphi^2 - \lambda^2}} \leq 1.0 \quad (4)$$

$$\varphi = 0.5(1 + \alpha(\lambda - 0.2) + \lambda^2) \quad (5)$$

124
125 It is noted that the f_k value is reduced by including $\sin \theta_1$ in the f_k function for X joints (see Eq. (2)) and by
126 incorporating the term of $(1/\sin \theta_1)^{0.5}$ in the λ_C equation (see Eq. (3)). This is because the research conducted by
127 Platt [18] showed that the effect of θ_1 on the resistance of a chord sidewall, in an RHS X joint with $\theta_1 < 90^\circ$, is
128 considerably smaller than being proportional to $1/\sin \theta_1$. Furthermore, a reduction factor of 0.8 (see Eq. (2)) (i.e., a
129 safety factor of 1.25) was adopted for RHS X joints to increase the safety margin for the X joints with higher chord
130 sidewall slenderness (h_0/t_0) which exhibit less-ductile failure.

131
132 Initially, no chord stress function (Q_f) was included for chord sidewall failure because the influence of small chord
133 stresses is insignificant. Later on, based on the research by Wardenier et al. [26], the following Q_f functions, which
134 are the same for $\beta=1.0$, were adopted for RHS T, Y and X joints [8, 25]:

$$Q_f = (1 - |n|)^{0.6-0.5\beta} \quad \text{for chord compression stress } (n < 0) \quad (6)$$

$$Q_f = (1 - |n|)^{0.1} \quad \text{for chord tension stress } (n \geq 0) \quad (7)$$

135 where n is the normal (longitudinal) stress ratio in the chord connecting face. The n value is taken as the sum of
136 the ratio of the chord axial force ($N_{0,Ed}$) to the chord axial yield capacity ($N_{pl,0,Rd}$) and the ratio of the chord bending
137 moment ($M_{0,Ed}$) to the chord plastic moment capacity ($M_{pl,0,Rd}$). Negative and positive n values denote chord
138 compression and tension stresses, respectively.

139
140 The resistance equations for chord sidewall failure in mild steel RHS X, T and Y joints under brace axial loading
141 have been extended for brace in-plane bending (see Eq. (8)) and for brace out-of-plane bending (see Eq. (9)) as
142 follows [8, 10, 25]:

$$M_{ip,1,Rd} = 0.5\chi_C f_{y0} t_0 (h_1 + 5t_0)^2 Q_f \quad (8)$$

$$M_{op,1,Rd} = \chi_C f_{y0} t_0 (b_0 - t_0) (h_1 + 5t_0) Q_f \quad (9)$$

143 which are conservative for $\theta_1 < 90^\circ$. For brace out-of-plane bending, it is presumed that the chord distortion failure
144 mode is prevented.

145

146 The aforementioned developments resulted in the resistance equations summarized in Table 1, which have been
147 adopted by recent international design codes and design guides [8, 10, 13, 25]. Up to 2013, the design
148 recommendations applied to a nominal yield stress (f_{y0}) of the finished hollow section up to 460 MPa, with the f_{y0}
149 value for design not exceeding 0.8 times the ultimate stress of the chord (f_{u0}). The stipulated joint resistances in
150 the design recommendations [8, 10, 13] were to be multiplied by a material factor (C_f) of 0.90 for $355 \text{ MPa} < f_{y0} \leq$
151 460 MPa . The most recent prEN 1993-1-8 [25] has proposed: $C_f = 1.00$ for $f_{y0} \leq 355 \text{ MPa}$, $C_f = 0.90$ for 355 MPa
152 $< f_{y0} \leq 460 \text{ MPa}$, $C_f = 0.86$ for $460 \text{ MPa} < f_{y0} \leq 550 \text{ MPa}$, and $C_f = 0.80$ for $550 \text{ MPa} < f_{y0} \leq 700 \text{ MPa}$.

153

154 2.3. Modifications to codified bearing-buckling model

155

156 2.3.1. Linearized buckling reduction factor proposed by Kuhn et al. [21]

157

158 Kuhn et al. [21] showed that the column buckling reduction factor ($\chi_{0.5}$) for mild steel cold-formed RHS decreases
159 in an approximately linear manner with increasing h_0/t_0 ratio up to 50. The $\chi_{0.5}$ value was obtained using a reduced
160 chord sidewall slenderness ($\lambda_{0.5}$), which was first suggested by Yu [19]:

$$\lambda_{0.5} = 0.5 \lambda_c \quad (10)$$

161 It is assumed that the chord sidewall is fixed along the longitudinal edges, and thus the $\lambda_{0.5}$ value is taken as half
162 of that adopted in Table 1. Kuhn et al. [21] proposed to express the buckling reduction factor as a linear function
163 of the h_0/t_0 ratio and also to include empirical terms of $(1/\sin\theta_1)^{0.5}$ and $(f_{y0}/350)^{0.5}$ to consider the effects of brace
164 angle and steel grade. These proposals resulted in the following linearized equation of the buckling reduction factor
165 for RHS X joints having $h_1/(h_0 \sin\theta_1) > 0.25$ [21]:

$$\chi_{Kuhn} = 1.15 - 0.013 \frac{h_0}{t_0} \sqrt{\frac{1}{\sin\theta_1}} \sqrt{\frac{f_{y0}}{350}} \leq 1.0 \quad (11)$$

166 For plate-to-RHS X joints and RHS-to-RHS X joints with $h_1/(h_0 \sin\theta_1) \leq 0.25$, $\chi_{Kuhn} = 1.0$ is proposed to be used
167 within the general validity range given in Table 1, and the resistance for chord sidewall failure in RHS X joints
168 under brace axial compression can be obtained from [21]:

$$N_{Kuhn} = \chi_{Kuhn} f_{y0} t_0 \left(\frac{2h_1}{\sin\theta_1} + 10t_0 \right) Q_f \quad (12)$$

169 It is noted that the term of $f_k t_0 / \sin\theta_1$ in Eq. (1) becomes $\chi f_{y0} t_0$ when substituting $f_k = \chi f_{y0} \sin\theta_1$ for RHS X joints.

170

171 The moment capacities for chord sidewall failure in RHS X joints under brace in-plane bending ($M_{ip,Kuhn}$) and
172 brace out-of-plane bending ($M_{op,Kuhn}$) may be obtained from Eqs. (8-9), but replacing χ_C with χ_{Kuhn} in Eq. (11).

173

174 2.3.2. η^* correction proposed by Wardenier [22]

175

176 Apart from using the reduced chord sidewall slenderness ($\lambda_{0.5}$) for RHS joints that are sufficiently restrained
 177 against out-of-plane movements, Wardenier [22] proposed to reconsider the effect of brace-to-chord height ratio
 178 ($\eta^*=h_1/h_0$). This is because the numerical results of Yu [19] and Lan et al. [23-24] show that full-width RHS X
 179 and T joints with higher η^* and $2\gamma^*$ ($=h_0/t_0$) ratios have a more-abrupt chord sidewall failure mode, i.e., the load-
 180 deformation curve exhibits a sharp drop in load after the peak load. Thus, it would be logical to increase at least
 181 the safety margin for RHS joints with a less-ductile failure mode.

182

183 Wardenier [22] proposed to include a correction function of $(h_1/h_0)^{-0.15}$ in the resistance equation (see Eq. (1)) in
 184 order to increase the safety margin for full-width RHS joints with a less-ductile failure mode. The modified
 185 resistance equations for RHS X joints with $\theta_1=90^\circ$ and under brace axial compression then become:

$$N_{\text{Ward}} = f_{k,\text{Ward}} t_0 (2h_1 + 10t_0) \left(\frac{h_0}{h_1} \right)^{0.15} Q_f \quad (13)$$

$$f_{k,\text{Ward}} = \chi_{\text{Ward}} f_{y0} \quad (14)$$

186 where χ_{Ward} is the buckling reduction factor obtained using the Eurocode buckling curve c and the chord sidewall
 187 slenderness ($\lambda_{0.5}$) or the linearized approximation, e.g., as proposed by Kuhn et al. [21] (see Eq. (11)). Using Eq.
 188 (13) would result in an equal or higher safety margin for the less-ductile RHS joints when compared with the more-
 189 ductile joints with low η^* and $2\gamma^*$ ratios.

190

191 The moment capacities for chord sidewall failure in RHS X joints under brace in-plane bending and brace out-of-
 192 plane bending may be obtained using Eqs. (8-9), but replacing χ_C with χ_{Ward} . It is also worth noting that the initial
 193 analyses conducted by Wardenier [22] indicate that the brace angle effect needs to be reconsidered.

194

195 **2.4. Representative analytical models**

196

197 *2.4.1. Four-hinge yield line model proposed by Yu [19]*

198

199 In the 1990s, Yu [19] conducted an extensive study on uniplanar and multiplanar RHS joints. A four-hinge yield
 200 line model (see Fig. 3) was proposed for chord sidewall failure in RHS-to-RHS X and T joints under brace axial
 201 compression, brace in-plane bending and brace out-of-plane bending. The corresponding resistance equation for
 202 mild steel RHS-to-RHS X and T joints, with $\theta_1=90^\circ$ and under brace axial compression, is as follows:

$$N_{\text{Yu}} = 4\chi_{0.5} (\sqrt{\gamma} + \eta) f_{y0} t_0^2 \quad (15)$$

203 where γ ($=b_0/2t_0$) is the chord width to twice chord wall thickness ratio, η ($=h_1/b_0$) is the brace height to chord
 204 width ratio, and $\chi_{0.5}$ is the buckling reduction factor determined by substituting $\lambda_{0.5}$ (see Eq. (10)) into Eqs. (4-5).
 205 The four-hinge yield line model assumes that the chord sidewalls are fixed along the longitudinal edges.

206

207 The moment capacity of chord sidewall failure in mild steel RHS-to-RHS T and X joints, with $\theta_1=90^\circ$ and loaded
 208 under brace in-plane bending, is given by:

$$M_{ip,Yu} = \chi_{ip,0.5} \left(2\sqrt{\gamma} + \gamma\eta + \frac{1}{2\eta} \right) f_{y0} t_0^2 h_1 \quad (16)$$

209 where $\chi_{ip,0.5}$ is the buckling reduction factor which equals 1.0 for $\eta \leq 1$ and for $1 < \eta \leq 2$, is determined by:

$$\chi_{ip,0.5} = 1 + (\eta - 1) \left(\frac{1}{\varphi + \sqrt{\varphi^2 - \lambda_{0.5}^2}} - 1 \right) \quad (17)$$

210 The moment capacity of chord sidewall failure in mild steel RHS-to-RHS T and X joints, with $\theta_1=90^\circ$ and loaded
211 under brace out-of-plane bending, is given by:

$$M_{op,Yu} = \chi_{0.5} \left(\sqrt{2(1+2\gamma)} + 2\gamma\eta \right) f_{y0} t_0^2 b_1 \quad (18)$$

212

213 2.4.2. Plate buckling model proposed by Becque and Cheng [20]

214

215 Becque and Cheng [20] proposed a plate buckling model conservatively assuming that the chord sidewall is pinned
216 along the longitudinal edges for chord sidewall failure in RHS-to-RHS X joints under brace axial compression.

217 The corresponding resistance equation is as follows:

$$N_{Becque} = 2.4 \chi_{Becque} f_{y0} h_1 t_0 \quad (19)$$

218 where χ_{Becque} is the buckling reduction factor obtained using the relevant buckling curve, e.g., according to Eqs.
219 (4-5); however, a modified imperfection factor $\alpha=0.08$ is recommended and the proposed chord sidewall
220 slenderness is as follows:

$$\lambda_{Becque} = \sqrt{\frac{P_y}{P_{cr}}} = \sqrt{\frac{2.4 f_{y0} h_1 t_0}{2 f_{cr,Becque} h_1 t_0}} \quad (20)$$

$$f_{cr,Becque} = 1.346 \frac{\pi^2 E}{12(1-\nu^2)} \frac{t_0^2}{h_0 h_1} \quad (21)$$

221 where E is the steel elastic modulus and ν is the Poisson ratio taken as 0.3.

222

223 It is noted that this design method is proposed to predict the initiation of buckling of the chord sidewall. This
224 buckling load can be considerably lower than the joint resistance determined by the peak load or the load at an
225 indentation limit of $3\%b_0$, whichever occurs at a smaller deformation, which is commonly adopted in other studies.
226 This design method is therefore not included in the subsequent evaluation.

227

228 2.4.3. Plate buckling model proposed by Lan et al. [23-24]

229

230 The restraint from the chord face and the brace to the chord sidewall is stronger than that of a pinned-end boundary
231 condition, but weaker than that of fixed edges. Lan et al. [23] proposed an analytical model of plate buckling for
232 chord sidewall failure in RHS-to-RHS X joints which can properly consider the restraint and utilize the strain
233 hardening of steel materials by using the continuous strength method. Later, Lan et al. [24] simplified the resistance
234 equations without considering the strain hardening for RHS-to-RHS X and T joints to reduce the computational
235 effort. It is noted that the strain hardening in high-strength steel is not pronounced. Fig. 4 shows the proposed plate
236 buckling model for RHS-to-RHS X and T joints with $\theta_1=90^\circ$.

237

238 The elastic buckling stress equation proposed for RHS-to-RHS T and X joints with $\theta_1=90^\circ$, which can properly
 239 consider the restraint from the chord face and the brace to the chord sidewall, is as follows [23-24]:

$$f_{cr,Lan} = 3.2 \frac{\pi^2 E}{12(1-\nu^2)} \left(\frac{t_0}{h_0} \right)^{1.96} \left(\frac{h_0}{h_1} \right)^{0.66} \quad (22)$$

240 The overall cross-section slenderness of the chord sidewall is defined by [23-24]:

$$\lambda_{Lan} = \sqrt{\frac{f_{y0}}{f_{cr,Lan}}} \quad (23)$$

241 which can be obtained by substituting Eq. (22), $E=210000$ MPa and $\nu=0.3$ into Eq. (23):

$$\lambda_{Lan} = 0.024 \left(\frac{h_0}{t_0} \right)^{0.98} \left(\frac{h_1}{h_0} \right)^{0.33} \sqrt{\frac{f_{y0}}{355}} \quad (24)$$

242 which can be conservatively approximated by rounding off the exponents:

$$\lambda_{Lan} = 0.024 \frac{h_0}{t_0} \left(\frac{h_1}{h_0} \right)^{0.3} \sqrt{\frac{f_{y0}}{355}} \quad (25)$$

243

244 The plate buckling reduction factor (χ_{Lan}) neglecting the strain hardening, which is based on the base curves
 245 proposed by Lan et al. [23], is as follows [24]:

$$\chi_{Lan} = \begin{cases} 1.0 & \text{for } \lambda_{Lan} \leq 0.6 \\ 0.8 \left(1 - \frac{0.2}{\lambda_{Lan}^{1.6}} \right) \frac{1}{\lambda_{Lan}^{1.6}} & \text{for } \lambda_{Lan} > 0.6 \end{cases} \quad (26)$$

246 The curve of the χ_{Lan} equation is relatively linear for $\chi_{Lan}>0.6$ and is herein suggested to be approximated by:

$$\chi_{Lan} = 1.39 - 0.67\lambda_{Lan} \leq 1.0 \quad (27)$$

247 which can be obtained by substituting Eq. (25) into Eq. (27) for steel grades up to S960 and ratios of b_0/t_0 and h_0/t_0
 248 up to 40:

$$\chi_{Lan} = 1.39 - 0.016 \frac{h_0}{t_0} \left(\frac{h_1}{h_0} \right)^{0.3} \sqrt{\frac{f_{y0}}{355}} \leq 1.0 \quad (28)$$

249 The linearized buckling reduction factor (see Eq. (28)) can produce conservative resistance prediction for RHS
 250 joints using higher steel grades in combination with ratios of b_0/t_0 and h_0/t_0 larger than 35, and thus the original
 251 Eqs. (25-26) are suggested for such cases.

252

253 The joint resistance (N_{Lan}) for chord sidewall failure in RHS-to-RHS T and X joints with $\theta_1=90^\circ$ can be obtained
 254 from [24]:

$$N_{Lan} = \chi_{Lan} f_{y0} t_0 (2h_1 + 10t_0) Q_f \quad (29)$$

255 The joint resistance for chord sidewall failure in RHS-to-RHS T, Y and X joints under brace in-plane bending and
 256 brace out-of-plane bending may be obtained from Eqs. (8-9), but replacing χ_C with χ_{Lan} in Eq. (28). The linearized
 257 Lan method using Eqs. (28-29) will be examined in the subsequent analyses.

258

259 2.5. Proposed design methods

260

261 2.5.1. General

262

263 Lan et al. [23-24, 27-28] evaluated the material effect on the resistance of fabricated RHS and CHS X and T joints
264 under brace axial compression and proposed the following equation for the material factor (C_f) to quantify the
265 resistance reduction, which was resulted from the material effect:

$$C_f = 1.1 - 62 f_{y0} / E \leq 1.0 \quad (30)$$

266 An equivalent C_f equation as a function of only f_{y0} is proposed in this study to maintain a uniform format for
267 equations:

$$C_f = 1.1 - 0.1 f_{y0} / 355 \leq 1.0 \quad (31)$$

268 The differences between the calculated C_f values using Eqs. (30-31) are found to be marginal. The derived C_f
269 values are 1.00, 0.97, 0.90, 0.85 and 0.83 for steel grades of S355, S460, S700, S900 and S960, respectively. The
270 corresponding rounded-off C_f values of 1.00, 0.95, 0.90, 0.85 and 0.80 may be used for chord sidewall failure
271 under brace compression loading, which are more optimistic than the general C_f values stipulated in prEN 1993-
272 1-8 [25]. Eq. (31) is incorporated in the proposed design methods mainly because significant material softening in
273 the heat-affected zone of high-strength steel can occur in practice and the effect of fabrication imperfections can
274 be more pronounced for chord sidewall failure in high-strength steel RHS joints (see Section 5).

275

276 The codified bearing-buckling method adopts various compensations for the brace angle effect in RHS X joints
277 by including $\sin\theta_1$ in f_k and $(1/\sin\theta_1)^{0.5}$ in λ_c (see Section 2.2). It is noted that the correction of $\sin\theta_1$ and safety
278 factor of 0.8 in f_k are not adopted for RHS T and Y joints (see Eq. (2)). This leads to inconsistencies for the design
279 of RHS X and T/Y joints. The brace angle effect for RHS X, T and Y joints is herein recommended to be
280 approximated by only using a function of $(1/\sin\theta_1)^{0.5}$ in the final resistance equation, which is in line with Davies
281 and Roodbaraky [17].

282

283 Wardenier [22] proposed to adopt a correction function of $(h_1/h_0)^{-0.15}$ for the design joint resistance (see Eq. (13)).
284 However, it is more suitable, e.g., for the loading case of brace axial tension, to include the term of $(h_1/h_0)^{-0.15}$ in
285 the f_k function and to impose an upper limit of f_{y0} for f_k values, and thus the design joint resistance can be limited
286 by the yield resistance.

287

288 The aforementioned proposed modifications result in the basic resistance equation for chord sidewall failure in
289 RHS X, T and Y joints under brace axial compression as follows:

$$N_p = C_f f_{k,p} t_0 (2h_1 + 10t_0) \sqrt{\frac{1}{\sin\theta_1}} Q_f \quad (32)$$

$$f_{k,p} = \chi_p \left(\frac{h_0}{h_1} \right)^{0.15} f_{y0} \leq f_{y0} \quad (33)$$

290 where C_f is the proposed material factor (see Eq. (31)), $t_0(2h_1+10t_0)$ is the bearing area taken in line with the
 291 bearing-buckling model, Q_f is the chord stress function (see Eqs. (6-7)), $(1/\sin\theta_1)^{0.5}$ is the brace angle function, $f_{k,P}$
 292 is the buckling stress of the chord sidewall and χ_P is the proposed buckling reduction factor. Two alternative
 293 methods are proposed to derive the χ_P values in this study.

294

295 2.5.2. Modified bearing-buckling method

296

297 Some code committees prefer, as currently used, a design method for chord sidewall failure which adopts the
 298 column buckling curve, in order to maintain consistency between the design of RHS joints and that of members.
 299 A format similar to the current set-up in the design codes and design guides was thus employed. The proposed
 300 modified bearing-buckling method in this study adopts a buckling curve c with $\alpha=0.49$ in EN 1993-1-1 [6] and a
 301 reduced chord sidewall slenderness of $\lambda_{0.5}$ (see Eq. (10)). The buckling reduction factor ($\chi_{P,M}$) can therefore be
 302 obtained from:

$$\chi_{P,M} = \frac{1}{\varphi_M + \sqrt{\varphi_M^2 - \lambda_{0.5}^2}} \leq 1.0 \quad (34)$$

$$\varphi_M = 0.5(1 + 0.49(\lambda_{0.5} - 0.2) + \lambda_{0.5}^2) \quad (35)$$

303 where $\chi_{P,M}$ is the proposed modified buckling reduction factor. It is noted that the codified bearing-buckling method
 304 and the four-hinge yield line model adopt different buckling curves according to the fabrication methods of cross-
 305 sections (e.g., cold-formed or hot-finished). However, a buckling curve c is herein suggested for all cross-sections
 306 to simplify the design process and to produce resistance predictions on the conservative side.

307

308 2.5.3. Lan-Kuhn method

309

310 The linearized Kuhn method is based on the combined bearing-buckling model with the chord sidewall assumed
 311 to be fixed along the longitudinal edges and local buckling covered by the strut buckling coefficient [5,29], whereas
 312 the linearized Lan method is based on a plate local buckling model. In reality, bearing governs for low h_0/t_0 ratios
 313 and local buckling dominates for higher h_0/t_0 ratios. Therefore, the Lan-Kuhn method using a linearized function
 314 of buckling reduction factor is proposed in this study.

315

316 The effect of h_1/h_0 ratio is not considered in the Kuhn method. The Lan method adopts a term of $(h_1/h_0)^{0.3}$ in the
 317 buckling reduction factor (see Eq. (28)) to quantify the effect, and this approach was initially considered for the
 318 Lan-Kuhn method. However, it was found that this could result in large deviations of the predicted resistances
 319 especially for $\eta^* \neq 1.0$ when compared with the proposed modified bearing-buckling method. It is noted that two
 320 alternative design methods should give comparable resistances. More detailed discussions can be found in
 321 Wardenier et al. [30]. Therefore, the influence of h_1/h_0 ratio is included in the buckling stress equation (see Eq.
 322 (33)), and the effect of θ_1 is considered in the basic resistance equation (see Eq. (32)) in this study. Only the effects
 323 of the h_0/t_0 ratio and f_{y0} are quantified in the proposed equation for the buckling reduction factor ($\chi_{P,LK}$):

$$\chi_{P,LK} = 1.12 - 0.012 \frac{h_0}{t_0} \sqrt{\frac{f_{y0}}{355}} \leq 1.0 \quad (36)$$

324

325 The joint resistance for chord sidewall failure in RHS X, T and Y joints under brace in-plane bending and brace
326 out-of-plane bending may be obtained from Eqs. (8-9), but replacing $\chi_C f_{y0}$ with $f_{k,P}$ in Eq. (33).

327

328 2.5.4. Comparison of the buckling reduction factors

329

330 Table 2 shows a comparison of $\chi_{P,LK}$, χ_{Kuhn} and χ_{Lan} with $\chi_{P,M}$ for $h_0/t_0 \leq 40$, $h_1/h_0 = 1.0$, $235 \text{ MPa} \leq f_{y0} \leq 960 \text{ MPa}$ and
331 $\theta_1 = 90^\circ$. The $\chi_{P,LK}/\chi_{P,M}$, $\chi_{Kuhn}/\chi_{P,M}$, and $\chi_{Lan}/\chi_{P,M}$ ratios equal the corresponding resistance ratios because $h_1/h_0 = 1.0$
332 and the same basic resistance equation (Eq. (1)) is adopted. It is shown that the differences between $\chi_{P,LK}$ and $\chi_{P,M}$
333 are minor with a maximum deviation of 4%. These two proposed design methods therefore give almost equivalent
334 resistances. The χ_{Kuhn} equation also produces excellent approximations of $\chi_{P,M}$ values for lower steel grades;
335 however, it is observed that the χ_{Kuhn} value deviates from the $\chi_{P,M}$ value for steel grades of S700 and higher in
336 combination with a high h_0/t_0 ratio. The maximum deviation is 4% for steel grades up to S700 and becomes 16%
337 for S960 which is on the conservative side. It is also shown that χ_{Lan} values are generally higher than $\chi_{P,M}$ values
338 with a maximum discrepancy of 20% because the Lan method is based on a plate buckling model and is not related
339 to the column buckling curves. It should be noted that the deviations of χ_{Kuhn} and χ_{Lan} values from $\chi_{P,M}$ values could
340 be larger for $h_1/h_0 < 1.0$ and $h_1/h_0 > 1.0$ because the effect of the h_1/h_0 ratio is not considered in χ_{Kuhn} for the Kuhn
341 method; however, it is considered in χ_{Lan} for the Lan method and in $f_{k,P}$ for the proposed modified bearing-buckling
342 method. More detailed information can be found in Wardenier et al. [30].

343

344 3. Evaluation of design methods for full-width RHS X and T joints under brace axial compression

345

346 3.1. General

347

348 A database of test and numerical results totalling 248 full-width RHS X joints under brace axial compression
349 reported in the literature was established. Results of plate-to-RHS X joints were analysed by Kuhn et al. [21] and
350 are not further considered in this study. The compiled results were adopted to evaluate the following six design
351 methods:

352 (1) The bearing-buckling method, but using the Eurocode buckling curve c and $\lambda_{0.5}$ with $N_{C,M}$ defined by Eqs. (1)
353 and (10)

354 (2) The Kuhn linearized method in Section 2.3.1 with N_{Kuhn} defined by Eqs. (11-12)

355 (3) The Yu four-hinge yield line method in Section 2.4.1 with N_{Yu} defined by Eqs. (10) and (15)

356 (4) The Lan plate buckling method using the linearized approach in Section 2.4.3 with N_{Lan} defined by Eqs. (28-
357 29)

358 (5) The bearing-buckling method, but using the Eurocode buckling curve c, $\lambda_{0.5}$ and $(\eta^*)^{-0.15}$ correction in Section
359 2.5.2 with $N_{P,M}$ defined by Eqs. (32-35)

360 (6) The Lan-Kuhn method using the linearized approach in Section 2.5.3 with $N_{P,LK}$ defined by Eqs. (32-33) and
361 (36)

362 The original equations in Section 2 are used in this study unless specified. The corresponding joint resistances
363 obtained using the six design methods ($N_{C,M}$, N_{Kuhn} , N_{Yu} , N_{Lan} , $N_{P,M}$ and $N_{P,LK}$) will be compared with the test and

364 numerical resistances (N_{1u}) in the subsequent sections. It should be noted that $N_{C,M}=N_{P,M}$ for $\eta^*=1.0$, and the effect
 365 of the η^* correction could be evaluated by comparing $N_{C,M}$ with $N_{P,M}$ for $\eta^*<1$ and $\eta^*>1$.

366

367 It should be noted that the safety factor of 1.25 for RHS X joints, adopted by the aforementioned design codes and
 368 design guides, was set to be unity in the assessment of the design methods. The Eurocode buckling curve c was
 369 conservatively used for all RHS joints, regardless of whether tests or numerical models used hot-finished or cold-
 370 formed hollow sections. In addition, RHS joints with $N_{1u}/N_y>1.1$, where N_y is the joint yield resistance, were
 371 excluded from the analyses because such data may not be realistic and could lead to a large scatter for the
 372 subsequent statistical analyses. The N_y values for all the design methods in this study is obtained from:

$$N_y = f_{y0}t_0(2h_1 + 10t_0)\sqrt{\frac{1}{\sin\theta_1}} \quad (37)$$

373 where the term $(1/\sin\theta_1)^{0.5}$ is adopted to consider the brace angle effect, in line with Davies and Roodbaraky [17],
 374 and the Q_f function is not incorporated. For this comparison, the omission of Q_f function is conservative, especially
 375 for large absolute values of chord stress ratio (n), as it leads to lower N_{1u}/N_y ratios.

376

377 3.2. Test results of RHS-to-RHS X joints

378

379 Table 3 summarises the compiled test results totalling 51 full-width RHS-to-RHS X-joints under brace axial
 380 compression. Source references for most tests are given in Kuhn et al. [21] and Fan [31]. Additional test results of
 381 high-strength steel RHS-to-RHS X joints reported by Feldmann et al. [32] and Pandey and Young [33] were also
 382 collated. It is shown that five RHS joints have resistances exceeding $1.1N_y$ and therefore only the remaining 46
 383 RHS-to-RHS X joints will be included in the subsequent analyses. The parameter ranges for the screened test
 384 database were $\beta=1.0$, $12.6\leq 2\gamma\leq 42.2$, $12.6\leq 2\gamma^*\leq 56.9$, $0.50\leq \eta\leq 2.47$, $0.60\leq \eta^*\leq 1.00$, $-0.87\leq n\leq 0$, $44^\circ\leq \theta_1\leq 90^\circ$ and 228
 385 $\text{MPa}\leq f_{y0}\leq 1080 \text{ MPa}$. Cold-formed and hot-finished RHS were covered.

386

387 The brace angle effect is re-evaluated against the test results of RHS-to-RHS X joints with varying brace angles
 388 in this study. Davies et al. [14] and Packer [15] found that the effect of brace angle on the resistance of full-width
 389 RHS X joints is smaller than being proportional to $1/\sin\theta_1$. Davies and Roodbaraky [17] reported that, for brace
 390 axial compression and tension, the enhancement of resistance for decreasing the brace angle could be more
 391 accurately quantified by a function of $(1/\sin\theta_1)^{0.5}$. Therefore, in the current codified design rules (see Table 1), the
 392 brace angle effect is, based on the initial investigations by Platt [18], minimised by various compensations in the
 393 chord sidewall slenderness (λ_c) and the buckling stress (f_k) for the X joints. It is noted that the term of $f_k t_0/\sin\theta_1$ in
 394 Eq. (1) becomes $\chi f_{y0} t_0$ when substituting $f_k = \chi f_{y0} \sin\theta_1$ for RHS X joints. The following two options are assessed
 395 against test results of 19 selected RHS-to-RHS X joints with $\theta_1\leq 90^\circ$:

396 (1) Using the codified term of $h_1/\sin\theta_1$ in the final resistance equation for $N_{C,M}$, N_{Kuhn} , N_{Lan} , $N_{P,M}$ and $N_{P,LK}$, and
 397 also including a term of $(1/\sin\theta_1)^{0.5}$ in $\lambda_{0.5}$ for $N_{C,M}$ and $N_{P,M}$ and in χ_{Kuhn} for N_{Kuhn} (see Table 4). Including a
 398 $1/\sin\theta_1$ term in the final resistance equation for N_{Yu} .

399 (2) Only incorporating a term of $(1/\sin\theta_1)^{0.5}$ in the final resistance equation for $N_{C,M}$, N_{Kuhn} , N_{Yu} , N_{Lan} , $N_{P,M}$, and
 400 $N_{P,LK}$ (see Table 5).

401

402 The material factor (C_f) was not used for all the statistical analyses summarised in Tables 4-5 because the variation
403 in yield stresses is small. The mean values of $N_{1u}/N_{C,M}$, N_{1u}/N_{Yu} , N_{1u}/N_{Kuhn} , N_{1u}/N_{Lan} , $N_{1u}/N_{P,M}$ and $N_{1u}/N_{P,LK}$ ratios
404 are 1.11, 1.02, 1.13, 0.89, 1.11 and 1.05, respectively, with corresponding coefficients of variation (CoVs) of 0.103,
405 0.103, 0.105, 0.101, 0.103 and 0.104 for the first approach (Table 4). However, for the second option, the mean
406 values of $N_{1u}/N_{C,M}$, N_{1u}/N_{Yu} , N_{1u}/N_{Kuhn} , N_{1u}/N_{Lan} , $N_{1u}/N_{P,M}$ and $N_{1u}/N_{P,LK}$ ratios are 1.12, 1.05, 1.13, 0.95, 1.12 and
407 1.12, respectively, with corresponding CoVs of 0.086, 0.079, 0.087, 0.082, 0.086 and 0.085 (Table 5). It is shown
408 that the CoV values of the various design methods for each option are close. The mean values for the second option
409 are slightly higher and the corresponding CoV values are about 20% lower when compared with those employing
410 the first solution. Therefore, only including a term of $(1/\sin\theta_1)^{0.5}$ in the final resistance equation, which is simpler
411 and yields more consistent resistance predictions, is recommended to account for the brace angle effect.

412

413 The chord stress effect was assessed against test results of eight available RHS-to-RHS X joints with $\theta_1=90^\circ$ and
414 varying chord stress ratios (n) summarised in Table 6. The codified Q_f function was adopted for all the design
415 methods in the statistical analyses and $C_f=1.0$ was used for all the mild steel X joints. The mean values of $N_{1u}/N_{C,M}$,
416 N_{1u}/N_{Yu} , N_{1u}/N_{Kuhn} , N_{1u}/N_{Lan} , $N_{1u}/N_{P,M}$ and $N_{1u}/N_{P,LK}$ ratios are 1.21, 1.13, 1.22, 1.03, 1.21 and 1.23, respectively,
417 with corresponding CoVs of 0.088, 0.088, 0.088, 0.089, 0.088 and 0.088. All the design methods yield almost the
418 same CoVs because only the chord stress ratio is different for each test series and all other parameters are nearly
419 the same. It is also observed that the resistance ratios, which generally exceed 1.0, increase with increasing absolute
420 value of n ratio because for high $|n|$ values the Q_f function adopts a conservative lower bound for the chord stress
421 effect.

422

423 The material effect was evaluated against the screened database of 46 RHS-to-RHS X joints in Tables 7-8. The
424 approach of only including a term of $(1/\sin\theta_1)^{0.5}$ in the final resistance equation was adopted for all design methods
425 to reveal the best performance of these methods. For the design methods without using the proposed C_f factor (see
426 Table 7), the mean values of $N_{1u}/N_{C,M}$, N_{1u}/N_{Yu} , N_{1u}/N_{Kuhn} , N_{1u}/N_{Lan} , $N_{1u}/N_{P,M}$ and $N_{1u}/N_{P,LK}$ ratios are 1.13, 1.05,
427 1.15, 0.97, 1.12 and 1.13, respectively, with corresponding CoVs of 0.098, 0.096, 0.098, 0.097, 0.097 and 0.097.
428 For the design methods using the C_f factor (see Table 8), the mean values of $N_{1u}/N_{C,M}$, N_{1u}/N_{Yu} , N_{1u}/N_{Kuhn} , N_{1u}/N_{Lan} ,
429 $N_{1u}/N_{P,M}$ and $N_{1u}/N_{P,LK}$ ratios are 1.17, 1.10, 1.20, 1.01, 1.17 and 1.17, respectively, with corresponding CoVs of
430 0.092, 0.086, 0.116, 0.116, 0.091 and 0.095. Thus, including the C_f factor reduces the CoVs except for the Kuhn
431 method and the Lan method, which both use the linearized approach. This is mainly because the Kuhn and Lan
432 methods are conservative for the S960 specimens tested by Pandey and Young [33].

433

434 The design methods without using the C_f factor often yield unconservative resistance predictions for the test
435 specimens with yield stresses higher than 900 MPa (see Table 7). In contrast, including the C_f factor in the design
436 methods leads to safe resistance predictions for all the high-strength steel test specimens (see Table 8). Thus, the
437 proposed C_f factor is suggested to consider the material effect. It is also shown that the Yu method and the proposed
438 modified bearing-buckling method give the lowest CoVs, and other design methods yield slightly higher CoVs. It
439 should be noted that the low resistance ratios of the X5-S960 specimen (see Table 7) may be attributed to the
440 material softening in the heat-affected zones and/or an insufficient weld size, as commented by Feldmann et al.

441 [32]. The test evidence for high-strength steel RHS joints remains limited, and more related test and numerical
442 investigations are needed to assess the material effect comprehensively.

443

444 Most of the RHS-to-RHS X joints in Tables 7-8 have $\eta^* \approx 1.0$ and there are only two X joints with small η^* values
445 of 0.60 and 0.75. Thus, the effect of including the η^* correction in the f_k function cannot be fully revealed in the
446 overall statistical analyses and has been checked in Section 3.3 using the numerical data.

447

448 3.3. Numerical results of RHS-to-RHS X joints

449

450 Table 9 summarises the collated numerical results totalling 173 RHS-to-RHS X joints with $\theta_1=90^\circ$ reported by Yu
451 [19] and Kuhn et al. [21]. It is shown that 42 RHS joints have resistances exceeding $1.1N_y$ and therefore only the
452 remaining 131 joints will be used in the analyses. The parameter ranges for the screened numerical database were
453 $\beta=1.0$, $10 \leq 2\gamma \leq 35$, $10 \leq 2\gamma^* \leq 35$, $0.25 \leq \eta \leq 2.00$, $0.21 \leq \eta^* \leq 2.50$, $-0.80 \leq n \leq 0.75$, $\theta_1=90^\circ$ and $f_{y0}=355$ and 398 MPa. Cold-
454 formed and hot-finished RHS are included. It is noted that all the RHS-to-RHS X joints had $\theta_1=90^\circ$ and thus the
455 brace angle effect cannot be evaluated. The C_f values for $f_{y0}=355$ and 398 MPa are 1.00 and 0.99, respectively,
456 thus the material effect is insignificant for these X joints. Nevertheless, the C_f factor was adopted for all the design
457 methods to allow for direct comparison.

458

459 The effect of the η^* ratio was examined against the numerical results of 22 selected RHS-to-RHS X joints with
460 $n=0$ and $0.42 \leq \eta^* \leq 2.50$ (see Table 10). The mean values of $N_{1u}/N_{C,M}$, N_{1u}/N_{Yu} , N_{1u}/N_{Kuhn} , N_{1u}/N_{Lan} , $N_{1u}/N_{P,M}$ and
461 $N_{1u}/N_{P,LK}$ ratios are 1.19, 1.11, 1.20, 1.05, 1.20 and 1.20, respectively, with corresponding CoVs of 0.105, 0.080,
462 0.106, 0.051, 0.064 and 0.059. It is shown that including the term of $(\eta^*)^{-0.15}$ in the $f_{k,P}$ (see Eq. (33)) of proposed
463 design methods can reduce the CoV by about 40% when compared with the bearing-buckling method ($N_{C,M}$) and
464 the Kuhn method (N_{Kuhn}) in which the η^* effect is not considered. Incorporating the term of $(\eta^*)^{0.3}$ in the buckling
465 reduction factor (see Eq. (28)) of the Lan method can also significantly reduce the CoV and the improvement is
466 slightly better than that of the proposed design methods. For the Lan-Kuhn model, including the $(\eta^*)^{0.3}$ term in
467 $\chi_{P,LK}$ (see Eq. (36)), as used in the Lan method, instead of using the $(\eta^*)^{-0.15}$ correction in $f_{k,P}$ (see Eq. (33)), slightly
468 increases the CoV for the joints in Table 10 from 0.059 to 0.062, and the deviations of $N_{P,LK}$ from $N_{P,M}$ become
469 larger up to 7%. Thus, including the proposed η^* correction in $f_{k,P}$ is suggested.

470

471 The chord stress effect was assessed against numerical results of 10 selected RHS-to-RHS X joints with varying
472 n ratios (see Table 11) reported by Yu [19]. The codified chord stress function (Q_f) was adopted for all the design
473 methods in the statistical analyses. The mean values of $N_{1u}/N_{C,M}$, N_{1u}/N_{Yu} , N_{1u}/N_{Kuhn} , N_{1u}/N_{Lan} , $N_{1u}/N_{P,M}$ and
474 $N_{1u}/N_{P,LK}$ ratios are 1.34, 1.25, 1.34, 1.15, 1.34 and 1.34, respectively, with corresponding CoVs of 0.074, 0.063,
475 0.073, 0.073, 0.074 and 0.064. The Yu and Lan-Kuhn methods yield the lowest CoVs and provide the most
476 consistent strength predictions. It is also found that the resistance ratios, which all exceed 1.0, increase with
477 increasing absolute values of the n ratio because the codified Q_f function employs a conservative lower bound for
478 the chord stress effect. It is noted that these conclusions also apply to the numerical data with varying n ratios
479 reported by Kuhn et al. [21] (see Table 12). Similar observations were reported by Kim et al. [34] for RHS X joints
480 with β ratio up to 1.0 and with $f_{y0}=324$ MPa and 798 MPa, and also by Lan et al. [23] for fabricated RHS X joints

481 with $f_{y0}=460, 690$ and 960 MPa. Thus, the need for new chord stress functions is not apparent, and Eqs. (6-7) can
482 be adopted.

483

484 Table 12 shows the results of statistical analyses for the evaluation of all the design methods against the screened
485 numerical database of 131 RHS-to-RHS X joints. The mean values of $N_{1u}/N_{C,M}$, N_{1u}/N_{Yu} , N_{1u}/N_{Kuhn} , N_{1u}/N_{Lan} ,
486 $N_{1u}/N_{P,M}$ and $N_{1u}/N_{P,LK}$ ratios are 1.24, 1.15, 1.25, 1.10, 1.23 and 1.24, respectively, with corresponding CoVs of
487 0.102, 0.082, 0.104, 0.061, 0.065 and 0.064. It is demonstrated that the Lan method and the proposed design
488 methods produce the lowest CoVs and thus most consistent resistance predictions.

489

490 **3.4. Summary for RHS-to-RHS X joints**

491

492 The overall statistical analyses for the test database (see Tables 7-8) show that the Yu method gives the lowest
493 CoVs; however, the differences with other design methods are small. The approach of only incorporating a term
494 of $(1/\sin\theta_1)^{0.5}$ in the final resistance equation can more accurately quantify the brace angle effect and is preferred.
495 Including the C_f factor reduces the CoVs for all the design methods except for the Kuhn method and the Lan
496 method. Incorporating the C_f factor in all the design methods can yield safe resistance predictions for high-strength
497 steel test specimens and is preferred; however, more experimental and numerical studies on high-strength steel
498 joints are needed to further confirm the proposed C_f factor. The codified chord stress function (Q_f) is more
499 conservative for larger absolute values of n ratio. It is noted that only two X joints had small η^* ratios in the test
500 database, and thus the evaluation of η^* effect is based on the numerical data.

501

502 The overall statistical analyses for the numerical database (see Table 12) show that the Lan method and the
503 proposed design methods (i.e., the proposed modified bearing-buckling method and the Lan-Kuhn method)
504 produce the lowest CoVs. It is demonstrated that the effect of the η^* ratio on the joint resistance is significant.
505 Including the η^* correction in the buckling reduction factor or the buckling stress equation results in more
506 consistent resistance predictions. Similar to the analyses for the test database, the codified Q_f function is observed
507 to be more conservative for large absolute values of the n ratio. The Q_f function can be adopted to consider the
508 chord stress effect. It is noted that the numerical database only covers mild steel and $\theta_1=90^\circ$; thus, the
509 corresponding effects of steel material and brace angle for RHS-to-RHS X joints cannot be examined.

510

511 It can be concluded that the proposed Lan-Kuhn method gives good correlations with the test data and excellent
512 correlations with the numerical results, and is better than the Kuhn and Lan methods. The proposed modified
513 bearing-buckling method produces nearly equivalent resistance predictions when compared with the proposed
514 Lan-Kuhn method. Thus, it can be adopted as an alternative design method which is in line with the current design
515 rules employing column buckling curves to determine the joint resistance. Although the Yu method is also very
516 accurate, the proposed design methods which give designers more insights into the structural behaviour of RHS
517 joints are easier to use and thus are recommended for RHS-to-RHS X joints. Figs. 5-6 illustrate the comparison of
518 the test and numerical resistances with those predicted by the proposed design methods, both using the C_f factor.

519

520 **3.5. RHS X joints with only one RHS brace welded to the chord**

521

522 Table 13 summarises the collated test results totalling 22 RHS X joints with an RHS brace welded to one side of
523 the chord and with the support of a block, a flat plate or a rigid solid base at the opposite side of the chord. It should
524 be noted that although these test specimens have the physical appearance of RHS T joints, the load transfer was
525 comparable to that of an X joint without shear in the chord, and thus these specimens were classified as RHS X
526 joints in line with ISO 14346 [8]. The experimental database consists of test results reported by Barentse [2] for a
527 welded flat plate support, plus Zhao [35], Pandey and Young [36] and Fan [31] for a rigid solid base. The smaller
528 brace width on either chord side was taken as h_1 in Table 13. It is noted that the test results of RHS X joints with
529 an unwelded block support reported by Poloni [37] were not included. This is because the chord cross-sections
530 used had large h_0/t_0 or b_0/t_0 ratios of 57, and hence were potentially sensitive to fabrication tolerances and deviations
531 in the test set-up. The chord wall slenderness is also out of the typical parameter ranges commonly adopted in
532 practice. The RHS X joints with $N_{1w}/N_y > 1.1$ were excluded from the statistical analyses.

533

534 For the compiled RHS X joints, Kuhn et al. [21] proposed three conditions of the chord sidewall end-restraint
535 along the chord length direction and corresponding chord sidewall slenderness as follows:

- 536 (a) Fixed-fixed: member or plate welded to two opposite chord sides, with a chord sidewall slenderness of $\lambda_{0.5}$.
537 (b) Fixed-pinned: member or plate welded to one chord side and unwelded to the opposite chord side, with a
538 chord sidewall slenderness of $\lambda_{0.7} = 1.4\lambda_{0.5}$.
539 (c) Pinned-pinned: plates or supports unwelded to two opposite chord sides, with a chord sidewall slenderness of
540 $\lambda_{1.0} = 2\lambda_{0.5}$.

541 According to this classification, the RHS X joints with an RHS brace welded to one chord side and with a plate
542 support welded to the opposite chord side, tested by Barentse [2], can be categorized as class a. Table 14 shows
543 that the mean values of $N_{1w}/N_{P,M}$ and $N_{1w}/N_{P,LK}$ ratios are 1.07 and 1.08, respectively, with corresponding CoVs of
544 0.054 and 0.059. It is demonstrated that the proposed design methods are applicable for these RHS X joints.

545

546 The remaining RHS X joints investigated by Zhao [35], Pandey and Young [36], and Fan [31] using a rigid solid
547 base can be grouped as class b. Thus, a chord sidewall slenderness of $1.4\lambda_{0.5}$ and a buckling curve c were used to
548 derive the buckling reduction factor ($\chi_{P,M1}$) and the joint resistance for the proposed modified bearing-buckling
549 method. For the proposed Lan-Kuhn method, the buckling reduction factor may be obtained from:

$$\chi_{P,LK1} = 1.12 - 0.017 \frac{h_0}{t_0} \sqrt{\frac{f_{y0}}{355}} \quad \text{with } h_0/t_0 \leq 40(355/f_{y0})^{0.5} \text{ but } \leq 40 \quad (38)$$

550 It is noted that the buckling reduction factor for RHS X joints in class b decreases non-linearly with increasing
551 h_0/t_0 ratio, for high yield stress and large chord sidewall slenderness. Thus, the validity of the approach of using
552 $1.4\lambda_{0.5}$ and the proposed linearized $\chi_{P,LK1}$ function of Eq. (38) (which can become considerably conservative) has
553 to be limited by $h_0/t_0 \leq 40(355/f_{y0})^{0.5}$ but ≤ 40 . The proposed h_0/t_0 limits are 40, 40, 35, 28, 25 and 24 for steel grades
554 of S235, S355, S460, S700, S900 and S960, respectively. Such limits are comparable to the class 3 limit specified
555 in the current EN 1993-1-1 [6], therefore the chord cross-section can be alternatively limited to class 3. This leaves
556 only one RHS X joint for S960, and the results of statistical analyses for the screened test database of class b are
557 shown in Table 15. The mean values of $N_{1w}/N_{P,M}$ and $N_{1w}/N_{P,LK}$ ratios are 1.16 and 1.18, respectively, with
558 corresponding CoVs of 0.130 and 0.141. It is shown that the proposed design methods provide conservative

559 resistance predictions. “RHS X joints” with members unwelded to two opposite chord sides in class c are not
560 examined in this study, but the chord sidewall slenderness of $\lambda_{1,0}$ suggested by Kuhn et al. [21] may be used.

561

562 **3.6. RHS T and Y joints**

563

564 Yu [19] conducted numerical simulations on chord sidewall failure in full-width RHS-to-RHS T joints. For the T
565 joints under brace axial compression, the global chord bending at the chord crown was eliminated by applying
566 compensating moments at the chord ends (i.e., $Q_f=1.0$). The resistance of one full-width RHS-to-RHS T joint with
567 $2\gamma=24$ was 1% higher than that of the comparable x11a specimen (see Table 9), and the same design rules were
568 proposed to be applied to RHS-to-RHS X and T joints. The aforementioned design recommendations developed
569 for RHS X joints are thus suggested for RHS T joints, which is also line with the current design codes and design
570 guides. The approach of only including a term of $(1/\sin\theta_1)^{0.5}$ in the final resistance equation is suggested for RHS
571 Y joints to consider the brace angle effect, which gives unified design rules for RHS X, T and Y joints.

572

573 Yu [19] also numerically examined the chord stress effect on RHS-to-RHS X and T joints with varying chord
574 sidewall slenderness and chord stress ratios. It was also shown that the effect of the bending moment could be
575 considered by the Q_f function. The plastic moment resistance ($M_{pl,0,Rd}$) for class 1 or 2 chord cross-sections and the
576 elastic moment resistance ($M_{el,0,Rd}$) for class 3 chord cross-sections could be adopted to calculate the chord stress
577 ratio. Such recommendations will be incorporated in the subsequent proposed design rules in Section 7.

578

579 **4. Discussion on full-width RHS X and T joints under brace axial tension**

580

581 Test data of full-width RHS X and T joints subjected to brace axial tension are available for mild steel and high-
582 strength steel in the literature; however, reanalyses of the test results are required. De Koning and Wardenier [38]
583 summarised the up-to-date test results for mild steel up to 1984 and compared the test resistances with those
584 obtained from the resistance equations for chord sidewall failure and brace failure given by Wardenier [29], which
585 are nearly identical to those in the current design codes and design guides. These test results confirm the suitability
586 of the codified design rules for steel grades up to and including S355.

587

588 Contradictory research findings have been reported for RHS joints in higher steel grades. For example, for S450
589 RHS-to-RHS X and T joints, Becque and Wilkinson [39] recommended the use of material factors for RHS joints
590 with non-ductile fracture failure modes. For the full-width X joints, brittle chord corner fracture and brace failure
591 were observed, both with low deformation capacity. In contrast, Björk and Saastamoinen [40] and Tuominen and
592 Björk [41] concluded, based on an assessment of the design equations in EN 1993-1-8 [7], that no material factors
593 are required for RHS-to-RHS X joints using S420 and S460 and the joints could be considered as being ductile.
594 Feldmann et al. [32] suggested material factors of 1.0, 0.90 and 0.80 for steel grades of S500, S700 and S960,
595 respectively. It is noted that the analyses conducted by Feldmann et al. [32] are based on a comparison of the test
596 resistances with the Eurocode design resistances, and no separate statistical analyses per failure mode were
597 conducted. The failure modes observed in tests sometimes deviated from those predicted by EN 1993-1-8 [7],
598 which incorporates different safety factors in the design equations for various failure modes.

599

600 It is noted that most of the tests have been carried out for RHS joints with square hollow section (SHS) brace and
601 chord having the same steel grade and wall thickness. Comparison of the resistance equations for the brace
602 effective width failure with those for chord sidewall failure in RHS joints with $\theta_1=90^\circ$ shows that the equations
603 then become rather similar. Furthermore, the material softening in the brace and chord resulting from welding
604 could vary and thus may alter the failure location. These factors explain the observed change in failure modes for
605 higher-strength steel joints.

606

607 Therefore, more detailed analyses of the aforementioned test results are needed for chord sidewall failure in full-
608 width RHS joints under brace axial tension. The resistance and deformation capacity per failure mode need to be
609 re-evaluated to ascertain whether, for the proposed design methods, lower resistance factors (ϕ) or higher safety
610 factors (γ_M) have to be applied for RHS joints using higher steel grades. Further, it is important that the steel
611 materials used for tests are representative of those in production specifications.

612

613 5. Discussion on RHS X and T joints under brace in-plane bending

614

615 Table 16 shows the compiled numerical results totalling eight full-width RHS-to-RHS X joints under brace in-
616 plane bending reported by Yu [19]. The numerical resistances ($M_{1u,ip}$) were compared with the yield resistances
617 ($M_{y,ip}$) obtained from:

$$M_{y,ip} = 0.5f_{y0}t_0(h_1 + 5t_0)^2 Q_f \quad (39)$$

618 It is shown that all joints, except for the x12ie2 specimen, reach the yield resistance ($M_{y,ip}$), and the resistance
619 ratios ($M_{1u,ip}/M_{y,ip}$) of all joints exceed 1.1 except for the specimens of x11ie2 and x12ie2.

620

621 The collated numerical results were adopted to evaluate the six design methods described in Section 3.1. The
622 codified resistance equation (see Eq. (8)) was used. However, the χ_C in Eq. (8) was replaced with $\chi_{0.5}$ for the
623 modified bearing-buckling method ($M_{C,M,ip}$), χ_{Kuhn} (see Eq. (11)) for the Kuhn method ($M_{Kuhn,ip}$), $\chi_{ip,0.5}$ (see Eq. (17))
624 for the Yu method ($M_{Yu,ip}$) and χ_{Lan} (see Eq. (28)) for the Lan method ($M_{Lan,ip}$). The term of $\chi_C f_{y0}$ in Eq. (8) was
625 replaced with $f_{k,P}$ (see Eq. (33)) for the proposed modified bearing-buckling method ($M_{P,M,ip}$) and the Lan-Kuhn
626 method ($M_{P,LK,ip}$). The Eurocode buckling curve c was conservatively used for all the RHS joints using hot-finished
627 hollow sections.

628

629 Table 17 summarises the results of the statistical analyses. It is shown that the Yu method gives lowest CoV of
630 0.072, and the CoVs of all other design methods are relatively large. However, it would be currently difficult to
631 draw conclusions with respect to the design methods. This is because the $M_{1u,ip}/M_{y,ip}$ ratios of most of the RHS-to-
632 RHS X joints are higher than 1.1. Additionally, for all the design methods, the plastic moment resistance, assuming
633 that the stress within the bearing length of (h_1+5t_0) all reaches the yield stress (f_{y0}), is used and the local buckling
634 effect is considered by the buckling reduction factor. This means that the strain at the outer part of the bearing
635 length would be considerably high, which may result in premature fracture failure for high-strength steel RHS
636 joints. More tests are needed to evaluate the suitability of these design methods for chord sidewall failure in higher-
637 strength steel RHS joints. It has to be examined whether the resistance of high-strength steel joints can be based

638 on a plastic stress distribution.

639

640 Yu [19] also reported that the resistances of six full-width RHS-to-RHS T joints under brace in-plane bending
641 were close to those of comparable RHS-to-RHS X joints (i.e., specimens of x10ie05, x10ie, x10ie2, x11ie2, x12i
642 and x12ie2 in Table 16) with a maximum positive deviation of 5%. Thus, it is recommended to adopt the same
643 resistance equations for full-width RHS-to-RHS X and T joints under brace in-plane bending.

644

645 Nagui [42] numerically examined the effects of chord sidewall convexity and thickness tolerance on full-width
646 RHS-to-RHS T joints under brace in-plane bending. All the T joints had $\eta^* = \eta = 1.0$. Table 18 shows a comparison
647 of the numerical resistances ($M_{1u,ip}$) with the predicted resistances ($M_{C,ip}$) obtained from Eq. (8) using $\chi_C = 1.0$. The
648 $M_{1u,ip}/M_{C,ip}$ ratio becomes smaller for higher steel grades indicating more significant effects of the fabrication
649 imperfections and more pronounced material effects. This further justifies the use of the material factor (C_f) which
650 could cover these effects for high-strength steel RHS joints.

651

652 **6. Discussion on RHS X and T joints under brace out-of-plane bending**

653

654 Table 19 tabulates the collated numerical results totalling eight full-width RHS-to-RHS X joints under brace out-
655 of-plane bending reported by Yu [19]. The numerical resistances ($M_{1u,op}$) were compared with the yield resistances
656 ($M_{y,op}$) derived from:

$$M_{y,op} = f_{y0} t_0 (b_0 - t_0) (h_1 + 5t_0) Q_f \quad (40)$$

657 It is shown that six joints reach the yield resistance ($M_{y,op}$), and the resistance ratios ($M_{1u,op}/M_{y,op}$) of three joints
658 exceed 1.1.

659

660 The numerical results were adopted to evaluate the six design methods described in Section 3.1. The codified
661 resistance equation (see Eq. (9)) was used. However, the χ_C in Eq. (9) was replaced with $\chi_{0.5}$ for the modified
662 bearing-buckling method ($M_{C,M,op}$), χ_{Kuhn} (see Eq. (11)) for the Kuhn method ($M_{Kuhn,op}$), $\chi_{0.5}$ for the Yu method
663 ($M_{Yu,op}$) and χ_{Lan} (see Eq. (28)) for the Lan method ($M_{Lan,op}$). The term of $\chi_C f_{y0}$ in Eq. (9) was replaced with $f_{k,P}$ (see
664 Eq. (33)) for the proposed modified bearing-buckling method ($M_{P,M,op}$) and the Lan-Kuhn method ($M_{P,LK,op}$). The
665 Eurocode buckling curve c was conservatively used for all the RHS joints using hot-finished hollow sections.
666 Table 20 shows that the proposed modified bearing-buckling method and the Lan-Kuhn method produce the lowest
667 CoVs of 0.054 and 0.046, respectively. However, similar to the discussion in Section 5, it is currently difficult to
668 draw generalised conclusions with respect to the design methods for the loading case of brace out-of-plane bending.
669 This is because the database is small with three X joints having $M_{1u,op}/M_{y,op} > 1.1$, and most of the X joints examined
670 reached the yield resistance. Fracture failure may occur due to the lower material ductility of high-strength steel.
671 More tests, in particular for chord sidewall failure in high-strength steel joints, are thus required.

672

673 Yu [19] also numerically studied full-width RHS-to-RHS T joints under brace out-of-plane bending. These joints
674 generally failed by distortion of the chord cross-section, and the corresponding joint resistance and stiffness largely
675 depend on the unstiffened chord length. If chord distortion is prevented, the same resistance equation can be
676 adopted for chord sidewall failure in RHS-to-RHS X and T joints under brace out-of-plane bending.

677

678 **7. Proposed design rules for RHS joints under brace axial compression**

679

680 More investigations on chord sidewall failure in high-strength steel RHS joints under brace axial tension, brace
681 in-plane bending and brace out-of-plane bending are needed to assess the design methods comprehensively.
682 Therefore, only design rules for chord sidewall failure in RHS X, T and Y joints under brace axial compression
683 are proposed herein. The numerical study conducted by Yu [19] shows that the resistances of RHS-to-RHS T joints
684 with $n=0$ are slightly higher than those of comparable RHS-to-RHS X joints. The approach of only including a
685 term of $(1/\sin\theta_1)^{0.5}$ in the final resistance equation is suggested for RHS X, T and Y joints in this study because of
686 the similar structural behaviour of these joints. Thus, the results of statistical analyses for X joints can be
687 considered to be also representative for T and Y joints. The proposed design methods using the recommended C_f
688 factor can provide conservative resistance predictions for full-width RHS X joints with only one RHS brace welded
689 to one chord side. For such RHS joints, the $\chi_{P,M}$ or $\chi_{P,LK}$ are appropriate for a welded plate (or similar) on the other
690 chord side (class a joints), and the proposed $\chi_{P,M1}$ or $\chi_{P,LK1}$ are suitable for an unwelded support on the other chord
691 side (class b joints). Hence, only the results of statistical analyses for RHS-to-RHS X joints under brace axial
692 compression were adopted to evaluate the mean resistance to the design resistance for the proposed design methods.

693

694 AISC 360-16 [43] stipulates a reliability index of 3.0 for ductile welded hollow section joints and often adopts the
695 simplified Eq. (41) from Ravindra and Galambos [44] to derive the resistance factor (ϕ). The beneficial overall
696 effects of variations of geometric parameters and material properties are neglected in the calibration.

$$\phi = (\text{Mean}) e^{(-0.55)(3.0)(\text{CoV})} \quad (41)$$

697 Table 8 shows that the mean values of the $N_{1w}/N_{P,M}$ and $N_{1w}/N_{P,LK}$ ratios are 1.17 and 1.17, respectively, with
698 corresponding CoVs of 0.091 and 0.095 for the evaluation against the test database. The corresponding obtained
699 ϕ factors are 1.01 and 1.00. For the assessment against the numerical database (see Table 12), the mean values of
700 the $N_{1w}/N_{P,M}$ and $N_{1w}/N_{P,LK}$ ratios are 1.23 and 1.24, respectively, with corresponding CoVs of 0.065 and 0.064.
701 The corresponding derived ϕ factors are 1.10 and 1.12. Thus, the smaller ϕ factors obtained from the evaluation
702 against the test results are governing and a rounded-off ϕ factor of 1.0 can be adopted for the two proposed design
703 methods. This indicates that the proposed modified bearing-buckling method and the Lan-Kuhn method produce
704 equivalent nominal and design resistances.

705

706 Both the proposed modified bearing-buckling method and the Lan-Kuhn method adopt chord sidewall slenderness
707 according to the conditions of chord sidewall end-resistant, an angle function of $(1/\sin\theta_1)^{0.5}$ in the final resistance
708 equation, an $(\eta^*)^{-0.15}$ correction in the f_k function and a buckling curve c or equivalent. Tables 21-22 summarise
709 the design rules using the two proposed design methods. It should be noted that for plate-to-RHS X joints and
710 RHS-to-RHS X joints with $h_1/h_0 \leq 0.25$, $f_k = f_{y0}$ is suggested in line with Kuhn et al. [21]; however, the nominal f_{y0}
711 values should not exceed 460 MPa due to the lack of test data for high-strength steel RHS joints.

712

713 **8. Conclusions**

714

715 This study deals with the design of chord sidewall failure in rectangular hollow section (RHS) X, T and Y joints.

716 Test and numerical results reported in the literature for chord sidewall failure in RHS joints were collated. A wide
717 range of geometric parameters, steel grades up to S960 and loading cases of brace axial loading, brace in-plane
718 bending and brace out-of-plane bending were investigated. The effects of brace-to-chord height ratio (η^*), brace
719 angle (θ_1), chord stress ratio (n) and steel grade were evaluated. The representative existing design approaches and
720 two proposed design methods were evaluated against the compiled test and numerical results. Further required
721 research on, in particular, high-strength steel RHS joints under brace axial tension, brace in-plane bending and
722 brace out-of-plane bending, was discussed. The conclusions for the loading case of brace axial compression are
723 summarised as follows:

724

725 (1) The effect of the η^* ratio on the joint resistance is pronounced and incorporating a correction term of $(\eta^*)^{-0.15}$
726 in the buckling stress function (f_k) significantly reduces the scatter of resistance predictions.

727

728 (2) The approach of only including a term of $(1/\sin\theta_1)^{0.5}$ in the final resistance equation can more accurately
729 quantify the effect of brace angles and is recommended.

730

731 (3) The current codified chord stress function (Q_f) becomes more conservative for large absolute values of the n
732 ratio and can be used to provide lower bound predictions for the chord stress effect.

733

734 (4) An equation for the material factor (C_f) is suggested to consider the material effect; the rounded-off C_f values
735 are 1.0, 0.95, 0.90, 0.85 and 0.80 for steel grades of S355, S460, S700, S900 and S960, respectively.

736

737 (5) The proposed modified bearing-buckling method and the simpler Lan-Kuhn method provide more consistent
738 resistance predictions when compared with the existing design methods.

739

740 (6) Tables 21-22, which are based on the two proposed alternative design methods, summarise the proposed
741 design rules for chord sidewall failure, which consider varying conditions of the chord sidewall end-restraint,
742 with a resulting resistance factor (ϕ) of 1.0.

743

744

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746

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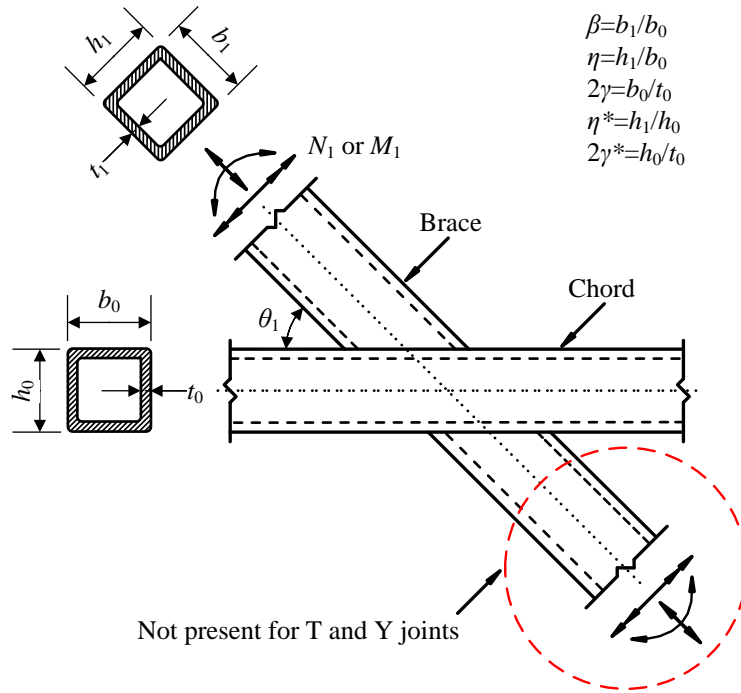


Fig. 1. Configurations and notations of RHS-to-RHS X, T and Y joints.

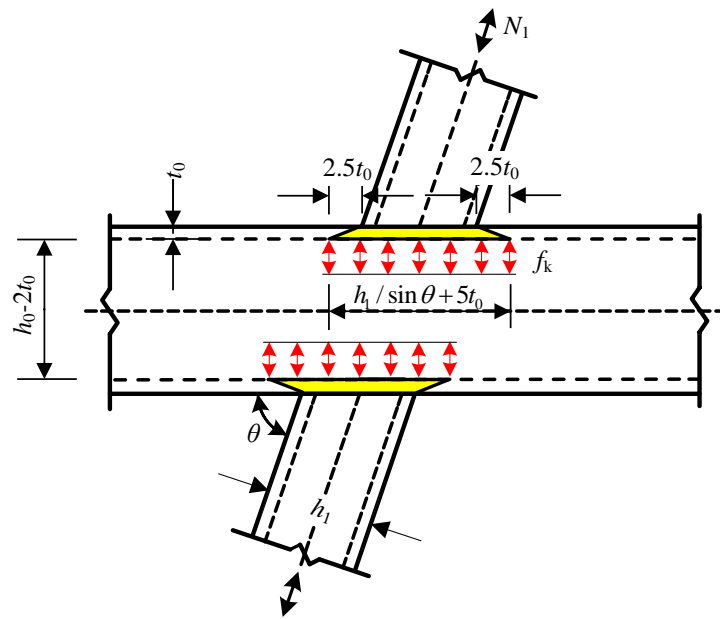


Fig. 2. Codified bearing-buckling model for chord side wall failure.

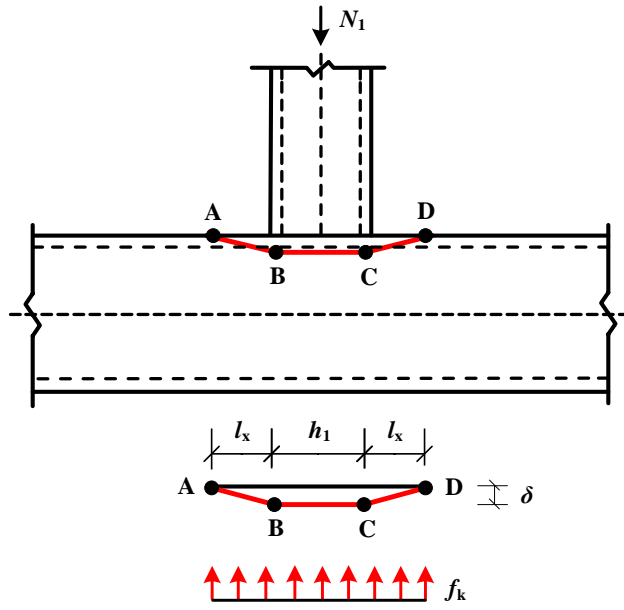


Fig. 3. Four-hinge yield line model proposed by Yu [19].

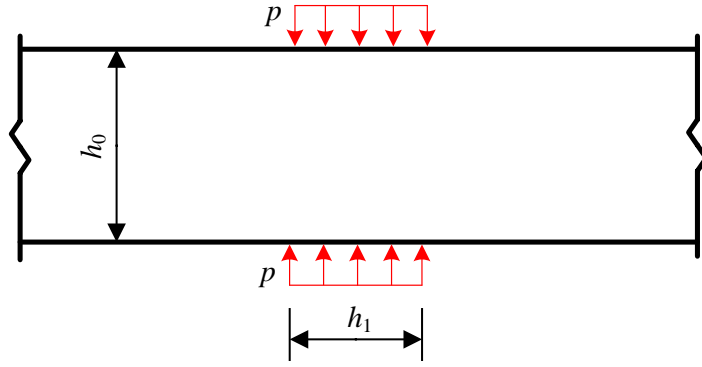
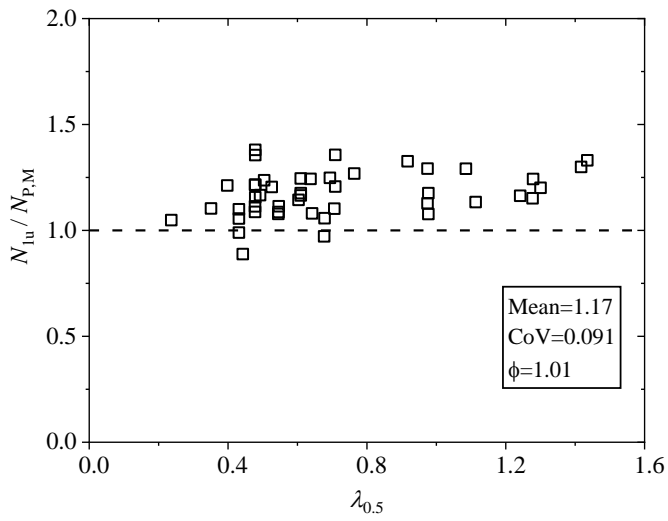
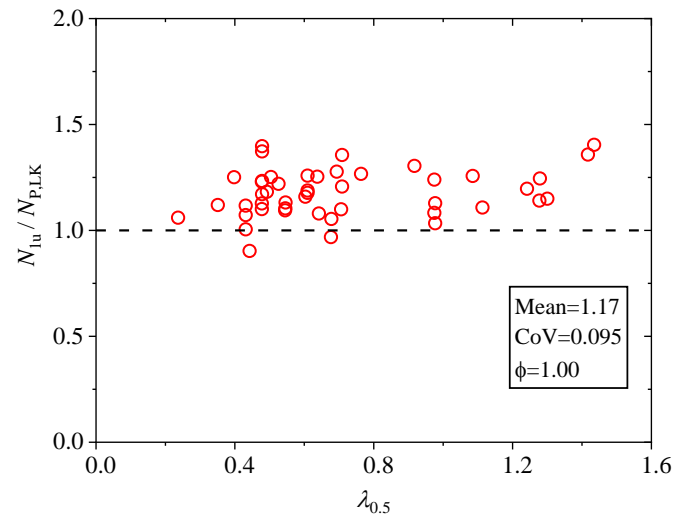


Fig. 4. Plate buckling model proposed by Lan et al. [23-24].

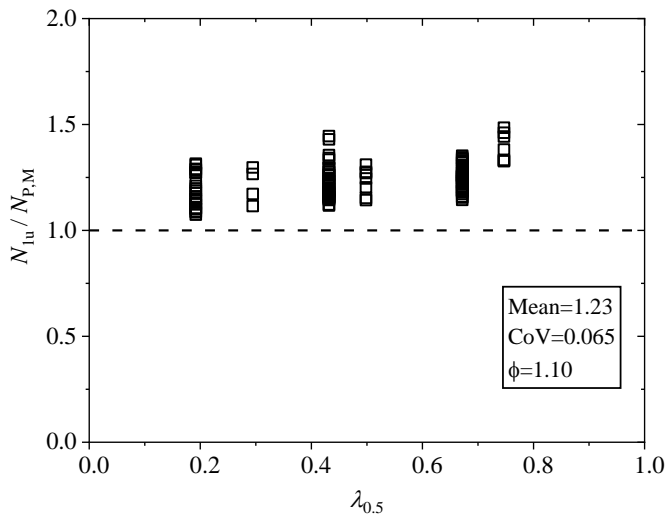


(a) Modified bearing-buckling method

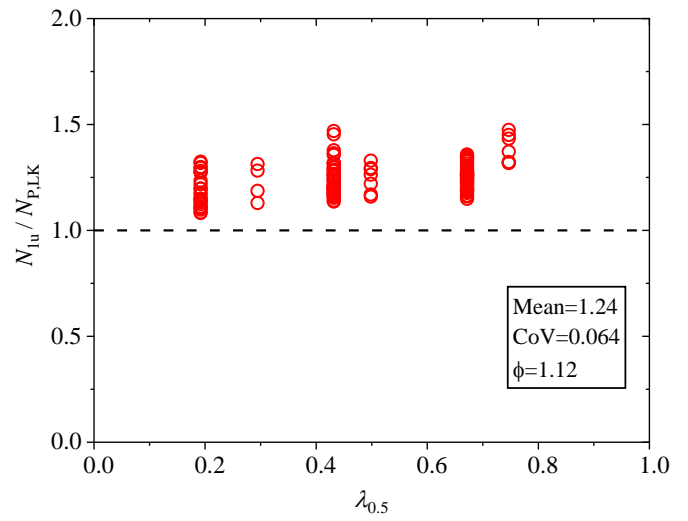


(b) Lan-Kuhn method

Fig. 5. Comparison of test resistances of 46 RHS-to-RHS joints under brace axial compression with those predicted by the proposed design methods, using the C_f factor.



(a) Modified bearing-buckling method



(b) Lan-Kuhn method

Fig. 6. Comparison of numerical resistances of 131 RHS-to-RHS joints under brace axial compression with those predicted by the proposed design methods, using the C_f factor.

Table 1

Codified design resistance for chord side wall failure in mild steel RHS joints with $\beta=1.0$ [8, 10, 13].

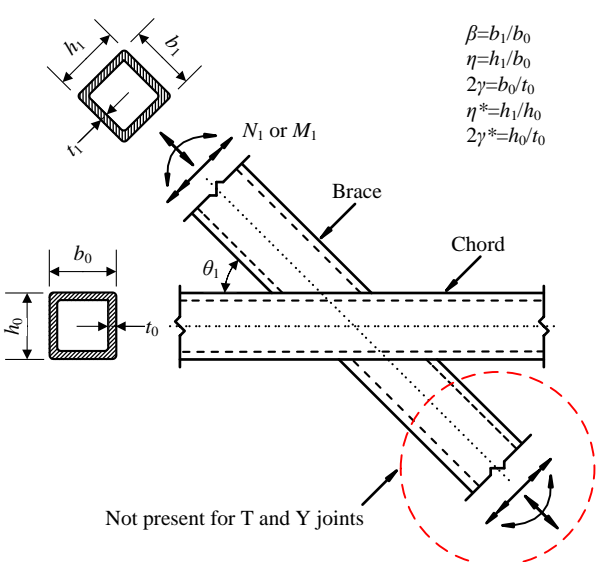
T, Y and X joints	Brace axial loading		
 <p style="text-align: right;"> $\beta=b_1/b_0$ $\eta=h_1/b_0$ $2\gamma=b_0/t_0$ $\eta^*=h_1/h_0$ $2\gamma^*=h_0/t_0$ </p> <p>Not present for T and Y joints</p>	$N_{1,Rd} = \frac{f_k t_0}{\sin \theta_1} \left(\frac{2h_1}{\sin \theta_1} + 10t_0 \right) Q_f$	<p>Tension: $f_k = f_{y0}$</p>	
		<p>Compression: $f_k = \chi_C f_{y0}$ for T and Y joints $f_k = 0.8 \chi_C f_{y0} \sin \theta_1$ for X joints</p>	
	<p>Brace in-plane bending</p>		<p>$f_k = f_{y0}$ for T and Y joints $f_k = 0.8 \chi_C f_{y0}$ for X joints</p>
	$M_{ip,1,Rd} = 0.5 f_k t_0 (h_1 + 5t_0)^2 Q_f$		
	<p>Brace out-of-plane bending</p>		<p>$f_k = \chi_C f_{y0}$ for T and Y joints $f_k = 0.8 \chi_C f_{y0}$ for X joints</p>
	$M_{op,1,Rd} = f_k t_0 (b_0 - t_0) (h_1 + 5t_0) Q_f$		
<p>Parameters</p>			
<p>where χ_C is the reduction factor for column buckling according to e.g., EN 1993-1-1 [6] using the relevant buckling curves and a normalised slenderness defined by:</p> $\lambda_c = \frac{3.46 \left(\frac{h_0}{t_0} - 2 \right) \sqrt{\frac{1}{\sin \theta_1}}}{\pi \sqrt{\frac{E}{f_{y0}}}}$			
<p>$Q_f = (1 - n)^{0.1}$ with n in connecting chord face</p> <p>$n = \frac{N_{0,Ed}}{N_{pl,0,Rd}} + \frac{M_{0,Ed}}{M_{pl,0,Rd}}$ for class 1 or 2 chord cross-sections under chord compression stress and for chord cross-sections under chord tension stress</p>			
<p>Validity ranges</p>			
<p>steel grades up to S355; $b_0/t_0 \leq 40$, $h_0/t_0 \leq 40$, the chord cross-section shall be class 1 or 2 for the chord under compression stress; $0.5 \leq h_0/b_0 \leq 2.0$; $\theta_1 \geq 30^\circ$.</p>			

Table 2Comparison of buckling reduction factors for RHS joints with $h_1/h_0=1.0$ and $\theta_1=90^\circ$.

f_{y0} (MPa)	h_0/t_0	$\chi_{P,M}$	$\chi_{P,LK}$	χ_{Kuhn}	χ_{Lan}	$\chi_{P,LK}/\chi_{P,M}$	$\chi_{Kuhn}/\chi_{P,M}$	$\chi_{Lan}/\chi_{P,M}$
960	10	0.95	0.92	0.93	1.00	0.97	0.98	1.05
	20	0.74	0.73	0.72	0.86	0.98	0.97	1.16
	30	0.52	0.53	0.50	0.60	1.02	0.98	1.16
	35	0.42	0.43	0.40	0.47	1.02	0.94	1.12
	40	0.34	0.33	0.29	0.34	0.96	0.84	0.98
700	10	0.97	0.95	0.97	1.00	0.98	0.99	1.03
	20	0.80	0.78	0.78	0.94	0.98	0.98	1.17
	30	0.61	0.61	0.60	0.72	1.01	0.99	1.18
	40	0.43	0.45	0.41	0.49	1.04	0.96	1.14
460	10	1.00	0.98	1.00	1.00	0.99	1.00	1.00
	20	0.86	0.85	0.85	1.00	0.98	0.99	1.16
	30	0.71	0.71	0.70	0.84	1.00	0.99	1.19
	40	0.55	0.57	0.55	0.66	1.04	1.00	1.20
355	10	1.00	1.00	1.00	1.00	1.00	1.00	1.00
	20	0.89	0.88	0.89	1.00	0.99	0.99	1.12
	30	0.77	0.76	0.76	0.91	0.99	0.99	1.19
	40	0.62	0.64	0.63	0.75	1.03	1.00	1.20
235	10	1.00	1.02	1.00	1.00	1.02	1.00	1.00
	20	0.93	0.92	0.94	1.00	0.99	1.00	1.07
	30	0.83	0.83	0.83	1.00	0.99	1.00	1.20
	40	0.72	0.73	0.72	0.87	1.01	1.00	1.20
Max					1.04	1.00	1.20	
Min					0.96	0.84	0.98	
Mean					1.00	0.98	1.12	
CoV					0.023	0.036	0.070	

Table 3

Collated test results totalling 51 RHS-to-RHS X joints under brace axial compression.

Researcher/year	Specimen	b_0 (mm)	h_0 (mm)	t_0 (mm)	b_1 (mm)	h_1 (mm)	t_1 (mm)	f_{y0} (MPa)	θ_1 (°)	n	η^*	$2\gamma^*$	N_{1u} (kN)	N_{1u}/N_y
Davies/1982	X(3)RR90	100.0	100.0	3.97	100.0	100.0	4.00	320	90	0	1.00	25.2	353	1.16
	X(4)RR45	100.0	100.0	3.93	100.0	100.0	4.00	320	45	0	1.00	25.4	372	1.04
Platt/1982	X-RR-90-A	100.0	100.2	4.20	100.0	100.0	4.00	432	90	0	1.00	23.9	391	0.89
	X-RR-60-A	99.8	100.1	4.20	100.0	100.0	4.00	432	60	0	1.00	23.8	410	0.87
	X-RR-45-A	100.1	100.0	4.20	100.0	100.0	4.00	432	45	0	1.00	23.8	450	0.86
	X-RR-90-B	98.7	100.0	4.00	100.0	50.0	5.00	311	90	0	0.50	25.0	209	1.20
	X-RR-60-B	98.9	100.0	4.10	100.0	50.0	5.00	311	60	0	0.50	24.4	218	1.13
	X-RR-45-B	99.2	100.0	4.00	100.0	50.0	5.00	311	45	0	0.50	25.0	244	1.18
	X-RR-90-C	250.0	251.1	6.50	250.0	250.0	6.30	237	90	0	1.00	38.6	680	0.78
	X-RR-60-C	250.4	250.7	6.50	250.0	250.0	6.30	237	60	0	1.00	38.6	672	0.72
X-RR-45-C	251.2	250.4	6.70	250.0	250.0	6.30	228	45	0	1.00	37.4	846	0.82	
Peksa/1982	10P	99.2	99.2	4.00	99.2	99.2	4.00	304	90	0	1.00	24.8	276	0.95
	11P	99.3	99.3	4.00	99.3	99.3	4.00	304	90	-0.43	1.00	24.8	271	0.93
	12P	99.3	99.3	4.00	99.3	99.3	4.00	304	90	-0.87	1.00	24.8	275	0.95
	13P	99.3	99.3	4.00	99.3	99.3	4.00	304	90	-0.87	1.00	24.8	280	0.97
	14P	99.2	99.2	4.00	99.2	99.2	4.00	304	90	-0.65	1.00	24.8	271	0.93
	15P	99.2	99.2	4.00	99.2	99.2	4.00	304	90	-0.22	1.00	24.8	263	0.91
Bettison/1982	5B	99.6	99.6	4.20	99.6	99.6	4.20	336	90	-0.71	1.00	23.7	313	0.92
	6B	99.8	99.8	4.10	99.8	99.8	4.10	336	90	-0.76	1.00	24.3	284	0.86
Poloni/1985	PWLR	102.4	252.7	4.44	102.4	252.7	4.44	388	90	0	1.00	56.9	438	0.46
Dixon/1983	DD1121	101.7	77.6	5.08	101.7	77.6	5.08	301	90	0	1.00	15.3	403	1.28
	DD1122	77.8	101.8	4.93	77.8	101.8	4.93	370	90	0	1.00	20.6	445	0.96
	DD1222	77.8	101.8	4.93	77.8	101.8	4.93	370	45	0	1.00	20.6	476	0.87
	DD1322	77.8	101.8	4.93	77.8	101.8	4.93	370	60	0	1.00	20.6	459	0.93
	DD2121	304.4	204.1	7.21	304.4	204.1	7.21	406	90	0	1.00	28.3	1315	0.93
	DD2122	204.1	304.4	7.21	204.1	304.4	7.21	406	90	0	1.00	42.2	1230	0.62
	DD2222	204.1	304.4	7.21	204.1	304.4	7.21	406	45	0	1.00	42.2	1675	0.71
	DD3121	203.2	153.6	4.83	203.2	153.6	4.83	392	90	0	1.00	31.8	649	0.97
	DD3122	153.6	203.2	4.83	153.6	203.2	4.83	412	90	0	1.00	42.1	530	0.59
	DD3221	203.2	153.6	4.83	203.2	153.6	4.83	392	44	0	1.00	31.8	693	0.86
	DD3222	153.6	203.2	4.83	153.6	203.2	4.83	412	44	0	1.00	42.1	694	0.64
	DD4123	254.1	254.1	9.35	254.1	254.1	9.35	406	90	0	1.00	27.2	2183	0.96
	DD4223	254.1	254.1	9.35	254.1	254.1	9.35	406	45	0	1.00	27.2	2429	0.90
DD4323	254.1	254.1	9.35	254.1	254.1	9.35	406	60	0	1.00	27.2	2215	0.90	
Cheng/2016	X1	100.5	100.3	2.92	100.2	100.3	2.73	330	90	0	1.00	34.3	176	0.79
	X2	100.4	100.1	3.84	100.4	100.2	3.69	330	90	0	1.00	26.1	302	1.00
	X3	100.3	99.8	4.89	100.1	99.9	4.70	400	90	0	1.00	20.4	373	0.77
	X4	99.6	99.6	5.80	99.8	99.7	5.46	370	90	0	1.00	17.2	560	1.01
	X5	99.9	99.7	7.92	100.1	99.6	7.68	345	90	0	1.00	12.6	783	1.03
	X6	149.8	250.0	5.00	150.1	150.1	4.76	463	90	0	0.60	50.0	409	0.50
	X7	150.2	150.2	5.86	150.5	150.4	5.86	451	90	0	1.00	25.6	828	0.87
	X9	300.0	400.0	7.92	300.3	300.3	7.97	481	90	0	0.75	50.5	1289	0.50
	Pandey/2020	X-100×50×4- 100×50×4	100.6	50.5	3.97	100.6	50.6	3.97	952	90	0	1.00	12.7	482
X-120×120×4- 120×120×4		121.6	121.7	3.93	121.4	121.8	3.92	971	90	0	1.00	31.0	567	0.53
X-140×140×4- 140×140×4		140.4	141.5	3.99	141.6	140.4	4.00	1008	90	0	0.99	35.5	484	0.37
X-120×120×3- 120×120×3		120.8	120.4	3.12	120.7	120.3	3.11	1038	90	0	1.00	38.6	317	0.36
X-80×80×4- 80×80×4		80.2	80.4	3.98	80.4	80.2	3.97	1004	90	0	1.00	20.2	595	0.74
X-120×120×4- 120×120×3		120.4	120.8	3.09	121.1	121.4	3.95	1038	90	0	1.00	39.1	318	0.36
Björk/2015	X5-S500	150.0	150.0	5.15	150.0	150.0	5.15	548	90	0	1.00	29.1	815	0.82
	X5-S700	150.0	150.0	5.06	150.0	150.0	5.06	762	90	0	1.00	29.6	935	0.69
	X5-S960	150.0	150.0	4.97	150.0	150.0	4.97	1080	90	0	1.00	30.2	808	0.43

Table 4Evaluation of design methods with the angle functions in $\lambda_{0.5}, f_k$ and the final resistance equations.

Specimen	f_{y0} (MPa)	θ_1 (°)	η^*	$2\gamma^*$	N_{1u} (kN)	$N_{1u}/N_{C,M}$	N_{1u}/N_{Y_u}	N_{1u}/N_{Kuhn}	N_{1u}/N_{Lan}	$N_{1u}/N_{P,M}$	$N_{1u}/N_{P,LK}$
X(4)RR45	320	45	1.00	25.4	372	1.17	1.04	1.19	0.92	1.17	1.11
X-RR-90-A	432	90	1.00	23.9	391	1.09	1.02	1.11	0.92	1.09	1.11
X-RR-60-A	432	60	1.00	23.8	410	1.04	0.96	1.06	0.85	1.04	1.03
X-RR-45-A	432	45	1.00	23.8	450	1.01	0.90	1.03	0.79	1.01	0.95
X-RR-90-C	237	90	1.00	38.6	680	1.06	0.97	1.06	0.88	1.06	1.05
X-RR-60-C	237	60	1.00	38.6	672	0.96	0.87	0.96	0.77	0.96	0.92
X-RR-45-C	228	45	1.00	37.4	846	1.04	0.93	1.05	0.79	1.04	0.94
DD1122	370	90	1.00	20.6	445	1.10	1.07	1.10	0.96	1.10	1.11
DD1322	370	60	1.00	20.6	459	1.02	0.97	1.04	0.88	1.02	1.02
DD1222	370	45	1.00	20.6	476	0.93	0.85	0.94	0.77	0.93	0.89
DD4123	406	90	1.00	27.2	2183	1.23	1.14	1.24	1.03	1.23	1.24
DD4323	406	60	1.00	27.2	2215	1.14	1.04	1.16	0.93	1.14	1.11
DD4223	406	45	1.00	27.2	2429	1.11	0.99	1.13	0.85	1.11	1.02
DD3121	392	90	1.00	31.8	649	1.34	1.21	1.35	1.13	1.34	1.34
DD3221	392	44	1.00	31.8	693	1.18	1.02	1.19	0.87	1.18	1.04
DD2122	406	90	1.00	42.2	1230	1.11	1.06	1.10	0.92	1.11	1.07
DD2222	406	45	1.00	42.2	1675	1.35	1.24	1.37	0.92	1.35	1.06
DD3122	412	90	1.00	42.1	530	1.06	1.00	1.05	0.88	1.06	1.02
DD3222	412	44	1.00	42.1	694	1.23	1.12	1.26	0.83	1.23	0.96
Mean						1.11	1.02	1.13	0.89	1.11	1.05
CoV						0.103	0.103	0.105	0.101	0.103	0.104

Note: The material factor (C_f) was not used for all design methods.

Table 5Evaluation of design methods only including a function of $(1/\sin\theta_1)^{0.5}$ in the final resistance equation.

Specimen	f_{y0} (MPa)	θ_1 (°)	η^*	$2\gamma^*$	N_{1u} (kN)	$N_{1u}/N_{C,M}$	N_{1u}/N_{Y_u}	N_{1u}/N_{Kuhn}	N_{1u}/N_{Lan}	$N_{1u}/N_{P,M}$	$N_{1u}/N_{P,LK}$
X(4)RR45	320	45	1.00	25.4	372	1.24	1.16	1.25	1.04	1.24	1.25
X-RR-90-A	432	90	1.00	23.9	391	1.09	1.02	1.11	0.92	1.09	1.11
X-RR-60-A	432	60	1.00	23.8	410	1.06	1.00	1.08	0.90	1.06	1.08
X-RR-45-A	432	45	1.00	23.8	450	1.05	0.99	1.07	0.89	1.05	1.07
X-RR-90-C	237	90	1.00	38.6	680	1.06	0.97	1.06	0.88	1.06	1.05
X-RR-60-C	237	60	1.00	38.6	672	0.97	0.89	0.97	0.81	0.97	0.97
X-RR-45-C	228	45	1.00	37.4	846	1.08	0.99	1.08	0.90	1.08	1.08
DD1122	370	90	1.00	20.6	445	1.10	1.07	1.10	0.96	1.10	1.11
DD1322	370	60	1.00	20.6	459	1.05	1.03	1.06	0.93	1.05	1.07
DD1222	370	45	1.00	20.6	476	0.99	0.96	0.99	0.87	0.99	1.00
DD4123	406	90	1.00	27.2	2183	1.23	1.14	1.24	1.03	1.23	1.24
DD4323	406	60	1.00	27.2	2215	1.16	1.08	1.17	0.98	1.16	1.17
DD4223	406	45	1.00	27.2	2429	1.15	1.07	1.16	0.97	1.15	1.16
DD3121	392	90	1.00	31.8	649	1.34	1.21	1.35	1.13	1.34	1.34
DD3221	392	44	1.00	31.8	693	1.19	1.07	1.21	1.00	1.19	1.19
DD2122	406	90	1.00	42.2	1230	1.11	1.06	1.10	0.92	1.11	1.07
DD2222	406	45	1.00	42.2	1675	1.27	1.21	1.26	1.06	1.27	1.22
DD3122	412	90	1.00	42.1	530	1.06	1.00	1.05	0.88	1.06	1.02
DD3222	412	44	1.00	42.1	694	1.16	1.09	1.15	0.96	1.16	1.11
Mean						1.12	1.05	1.13	0.95	1.12	1.12
CoV						0.086	0.079	0.087	0.082	0.086	0.085

Note: The material factor (C_f) was not used for all design methods.

Table 6

Evaluation of the chord stress effect against test results of eight RHS-to-RHS X joints.

Specimen	f_{y0} (MPa)	θ_1 (°)	n	η^*	N_{1u} (kN)	$N_{1u}/N_{C,M}$	N_{1u}/N_{Y_u}	N_{1u}/N_{Kuhn}	N_{1u}/N_{Lan}	$N_{1u}/N_{P,M}$	$N_{1u}/N_{P,LK}$
10P	304	90	0	1.00	276	1.11	1.04	1.12	0.95	1.11	1.13
11P	304	90	-0.43	1.00	271	1.16	1.08	1.16	0.99	1.16	1.17
12P	304	90	-0.87	1.00	275	1.36	1.27	1.36	1.16	1.36	1.37
13P	304	90	-0.87	1.00	280	1.38	1.29	1.39	1.18	1.38	1.40
14P	304	90	-0.65	1.00	271	1.21	1.14	1.22	1.04	1.21	1.23
15P	304	90	-0.22	1.00	263	1.09	1.02	1.09	0.93	1.09	1.10
5B	336	90	-0.71	1.00	313	1.22	1.14	1.23	1.04	1.22	1.23
6B	336	90	-0.76	1.00	284	1.17	1.09	1.18	0.99	1.17	1.18
Mean						1.21	1.13	1.22	1.03	1.21	1.23
CoV						0.088	0.088	0.088	0.089	0.088	0.088

Note: The codified chord stress function (Q_f) was adopted for all design methods and the material factor (C_f) equals 1.0 for all the joints.

Table 7Evaluation of design methods, without using the C_f factor, against test results of 46 screened RHS-to-RHS X joints under brace axial compression.

Specimen	f_{y0} (MPa)	θ_1 (°)	n	η^*	N_{1u} (kN)	$N_{1u}/N_{C,M}$	N_{1u}/N_{Yu}	N_{1u}/N_{Kuhn}	N_{1u}/N_{Lan}	$N_{1u}/N_{P,M}$	$N_{1u}/N_{P,LK}$
X(4)RR45	320	45	0	1.00	372	1.24	1.16	1.25	1.04	1.24	1.25
X-RR-90-A	432	90	0	1.00	391	1.09	1.02	1.11	0.92	1.09	1.11
X-RR-60-A	432	60	0	1.00	410	1.06	1.00	1.08	0.90	1.06	1.08
X-RR-45-A	432	45	0	1.00	450	1.05	0.99	1.07	0.89	1.05	1.07
X-RR-90-C	237	90	0	1.00	680	1.06	0.97	1.06	0.88	1.06	1.05
X-RR-60-C	237	60	0	1.00	672	0.97	0.89	0.97	0.81	0.97	0.97
X-RR-45-C	228	45	0	1.00	846	1.08	0.99	1.08	0.90	1.08	1.08
10P	304	90	0	1.00	276	1.11	1.04	1.12	0.95	1.11	1.13
11P	304	90	-0.43	1.00	271	1.16	1.08	1.16	0.99	1.16	1.17
12P	304	90	-0.87	1.00	275	1.36	1.27	1.36	1.16	1.36	1.37
13P	304	90	-0.87	1.00	280	1.38	1.29	1.39	1.18	1.38	1.40
14P	304	90	-0.65	1.00	271	1.21	1.14	1.22	1.04	1.21	1.23
15P	304	90	-0.22	1.00	263	1.09	1.02	1.09	0.93	1.09	1.10
5B	336	90	-0.71	1.00	313	1.22	1.14	1.23	1.04	1.22	1.23
6B	336	90	-0.76	1.00	284	1.17	1.09	1.18	0.99	1.17	1.18
PWLR	388	90	0	1.00	438	1.19	1.16	1.25	1.06	1.19	1.14
DD1122	370	90	0	1.00	445	1.09	1.07	1.10	0.96	1.09	1.11
DD1222	370	45	0	1.00	476	0.99	0.96	0.99	0.87	0.99	1.00
DD1322	370	60	0	1.00	459	1.05	1.03	1.06	0.93	1.05	1.07
DD2121	406	90	0	1.00	1315	1.23	1.09	1.24	1.03	1.23	1.24
DD2122	406	90	0	1.00	1230	1.11	1.06	1.10	0.92	1.11	1.07
DD2222	406	45	0	1.00	1675	1.27	1.21	1.26	1.06	1.27	1.22
DD3121	392	90	0	1.00	649	1.34	1.21	1.35	1.13	1.34	1.34
DD3122	412	90	0	1.00	530	1.06	1.00	1.05	0.88	1.06	1.02
DD3221	392	44	0	1.00	693	1.19	1.07	1.21	1.00	1.19	1.19
DD3222	412	44	0	1.00	694	1.16	1.09	1.15	0.96	1.16	1.11
DD4123	406	90	0	1.00	2183	1.23	1.14	1.24	1.03	1.23	1.24
DD4223	406	45	0	1.00	2429	1.15	1.07	1.16	0.97	1.15	1.16
DD4323	406	60	0	1.00	2215	1.16	1.08	1.17	0.98	1.16	1.17
X1	330	90	0	1.00	176	1.10	1.02	1.11	0.92	1.10	1.10
X2	330	90	0	1.00	302	1.20	1.12	1.22	1.01	1.20	1.22
X3	400	90	0	1.00	373	0.88	0.83	0.89	0.77	0.88	0.89
X4	370	90	0	1.00	560	1.10	1.06	1.10	1.01	1.10	1.11
X5	345	90	0	1.00	783	1.05	1.05	1.04	1.03	1.05	1.06
X6	463	90	0	0.60	409	1.22	1.13	1.25	0.83	1.13	1.07
X7	451	90	0	1.00	828	1.11	1.04	1.13	0.94	1.11	1.13
X9	481	90	0	0.75	1289	1.25	1.15	1.31	0.94	1.20	1.15
X-100×50×4- 100×50×4	952	90	0	1.00	482	1.01	0.90	1.03	0.91	1.01	1.04
X-120×120×4- 120×120×4	971	90	0	1.00	567	1.07	0.99	1.10	0.92	1.07	1.04
X-140×140×4- 140×140×4	1008	90	0	0.99	484	0.94	0.87	1.02	0.86	0.94	0.93
X-120×120×3- 120×120×3	1038	90	0	1.00	317	1.05	0.97	1.26	1.08	1.05	1.10
X-80×80×4- 80×80×4	1004	90	0	1.00	595	1.02	0.97	1.05	0.88	1.02	1.04
X-120×120×4- 120×120×3	1038	90	0	1.00	318	1.07	0.99	1.31	1.13	1.07	1.13
X5-S500	548	90	0	1.00	815	1.20	1.11	1.21	1.01	1.20	1.20
X5-S700	762	90	0	1.00	935	1.17	1.09	1.19	1.00	1.17	1.16
X5-S960	1080	90	0	1.00	808	0.90	0.84	0.93	0.79	0.90	0.88
Mean						1.13	1.05	1.15	0.97	1.12	1.13
CoV						0.098	0.096	0.098	0.097	0.097	0.097

Note: The approach of only including a term of $(1/\sin\theta_1)^{0.5}$ in the final resistance equation was adopted for all design methods.

Table 8Evaluation of design methods, using the C_f factor, against test results of 46 screened RHS-to-RHS X joints under brace axial compression.

Specimen	f_{y0} (MPa)	θ_1 (°)	n	η^*	N_{1u} (kN)	$N_{1u}/N_{C,M}$	N_{1u}/N_{Y_u}	N_{1u}/N_{Kuhn}	N_{1u}/N_{Lan}	$N_{1u}/N_{P,M}$	$N_{1u}/N_{P,LK}$
X(4)RR45	320	45	0	1.00	372	1.24	1.16	1.25	1.04	1.24	1.25
X-RR-90-A	432	90	0	1.00	391	1.11	1.05	1.13	0.94	1.11	1.13
X-RR-60-A	432	60	0	1.00	410	1.09	1.02	1.10	0.92	1.09	1.10
X-RR-45-A	432	45	0	1.00	450	1.08	1.01	1.09	0.91	1.08	1.09
X-RR-90-C	237	90	0	1.00	680	1.06	0.97	1.06	0.88	1.06	1.05
X-RR-60-C	237	60	0	1.00	672	0.97	0.89	0.97	0.81	0.97	0.97
X-RR-45-C	228	45	0	1.00	846	1.08	0.99	1.08	0.90	1.08	1.08
10P	304	90	0	1.00	276	1.11	1.04	1.12	0.95	1.11	1.13
11P	304	90	-0.43	1.00	271	1.16	1.08	1.16	0.99	1.16	1.17
12P	304	90	-0.87	1.00	275	1.36	1.27	1.36	1.16	1.36	1.37
13P	304	90	-0.87	1.00	280	1.38	1.29	1.39	1.18	1.38	1.40
14P	304	90	-0.65	1.00	271	1.21	1.14	1.22	1.04	1.21	1.23
15P	304	90	-0.22	1.00	263	1.09	1.02	1.09	0.93	1.09	1.10
5B	336	90	-0.71	1.00	313	1.22	1.14	1.23	1.04	1.22	1.23
6B	336	90	-0.76	1.00	284	1.17	1.09	1.18	0.99	1.17	1.18
PWLR	388	90	0	1.00	438	1.20	1.17	1.26	1.07	1.20	1.15
DD1122	370	90	0	1.00	445	1.10	1.07	1.11	0.97	1.10	1.12
DD1222	370	45	0	1.00	476	0.99	0.97	1.00	0.87	0.99	1.00
DD1322	370	60	0	1.00	459	1.06	1.03	1.06	0.93	1.06	1.07
DD2121	406	90	0	1.00	1315	1.24	1.10	1.26	1.05	1.24	1.25
DD2122	406	90	0	1.00	1230	1.13	1.07	1.12	0.94	1.13	1.08
DD2222	406	45	0	1.00	1675	1.29	1.23	1.28	1.07	1.29	1.24
DD3121	392	90	0	1.00	649	1.36	1.22	1.37	1.14	1.36	1.36
DD3122	412	90	0	1.00	530	1.08	1.01	1.07	0.90	1.08	1.03
DD3221	392	44	0	1.00	693	1.21	1.08	1.22	1.01	1.21	1.21
DD3222	412	44	0	1.00	694	1.18	1.11	1.17	0.98	1.18	1.13
DD4123	406	90	0	1.00	2183	1.24	1.16	1.26	1.05	1.24	1.26
DD4223	406	45	0	1.00	2429	1.16	1.08	1.18	0.98	1.16	1.18
DD4323	406	60	0	1.00	2215	1.18	1.09	1.19	0.99	1.18	1.19
X1	330	90	0	1.00	176	1.10	1.02	1.11	0.92	1.10	1.10
X2	330	90	0	1.00	302	1.20	1.12	1.22	1.01	1.20	1.22
X3	400	90	0	1.00	373	0.89	0.84	0.90	0.78	0.89	0.90
X4	370	90	0	1.00	560	1.10	1.06	1.11	1.02	1.10	1.12
X5	345	90	0	1.00	783	1.05	1.05	1.04	1.03	1.05	1.06
X6	463	90	0	0.60	409	1.26	1.16	1.29	0.86	1.16	1.11
X7	451	90	0	1.00	828	1.14	1.07	1.16	0.97	1.14	1.16
X9	481	90	0	0.75	1289	1.30	1.19	1.36	0.98	1.24	1.19
X-100×50×4- 100×50×4	952	90	0	1.00	482	1.21	1.08	1.24	1.09	1.21	1.25
X-120×120×4- 120×120×4	971	90	0	1.00	567	1.29	1.20	1.32	1.11	1.29	1.26
X-140×140×4- 140×140×4	1008	90	0	0.99	484	1.15	1.06	1.25	1.05	1.15	1.14
X-120×120×3- 120×120×3	1038	90	0	1.00	317	1.30	1.20	1.56	1.33	1.30	1.36
X-80×80×4- 80×80×4	1004	90	0	1.00	595	1.25	1.18	1.29	1.07	1.25	1.28
X-120×120×4- 120×120×3	1038	90	0	1.00	318	1.33	1.22	1.62	1.40	1.33	1.41
X5-S500	548	90	0	1.00	815	1.27	1.18	1.28	1.07	1.27	1.27
X5-S700	762	90	0	1.00	935	1.33	1.23	1.34	1.12	1.33	1.30
X5-S960	1080	90	0	1.00	808	1.13	1.05	1.17	0.99	1.13	1.11
Mean						1.17	1.10	1.20	1.01	1.17	1.17
CoV						0.092	0.086	0.116	0.116	0.091	0.095

Note: The approach of only including a term of $(1/\sin\theta_1)^{0.5}$ in the final resistance equation was adopted for all design methods.

Table 9

Collated numerical results totalling 173 RHS-to-RHS X joints under brace axial compression.

Researcher/year	Specimen	b_0 (mm)	h_0 (mm)	t_0 (mm)	b_1 (mm)	h_1 (mm)	t_1 (mm)	f_{y0} (MPa)	θ_1 (°)	n	η^*	$2\gamma^*$	N_{1u} (kN)	N_{1u}/N_y
Yu/1997	x10ae05	150	150	10.00	150	75	10.00	355	90	0	0.50	15	1019	1.15
	x10ae	150	150	10.00	150	150	10.00	355	90	0	1.00	15	1507	1.06
	x10ae2	150	150	10.00	150	300	10.00	355	90	0	2.00	15	2498	1.01
	x11ae05	150	150	6.25	150	75	6.25	355	90	0	0.50	24	509	1.08
	x11a	150	150	6.25	150	150	6.25	355	90	0	1.00	24	776	0.96
	x11ae2	150	150	6.25	150	300	6.25	355	90	0	2.00	24	1427	0.97
	x12ae05	150	150	4.29	150	75	4.29	355	90	0	0.50	35	302	1.03
	x12ae	150	150	4.29	150	150	4.29	355	90	0	1.00	35	482	0.92
	x12ae2	150	150	4.29	150	300	4.29	355	90	0	2.00	35	816	0.83
	x10a-0	150	150	10.00	150	150	10.00	355	90	0	1.00	15	1611	1.13
	x10a-0.4	150	150	10.00	150	150	10.00	355	90	-0.40	1.00	15	1587	1.12
	x10a-0.6	150	150	10.00	150	150	10.00	355	90	-0.60	1.00	15	1562	1.10
	x10a-0.8	150	150	10.00	150	150	10.00	355	90	-0.80	1.00	15	1492	1.05
	x11a-0	150	150	6.25	150	150	6.25	355	90	0	1.00	24	817	1.02
	x11a-0.4	150	150	6.25	150	150	6.25	355	90	-0.40	1.00	24	803	1.00
	x11a-0.6	150	150	6.25	150	150	6.25	355	90	-0.60	1.00	24	787	0.98
	x11a-0.8	150	150	6.25	150	150	6.25	355	90	-0.80	1.00	24	757	0.94
	x12a-0	150	150	4.29	150	150	4.29	355	90	0	1.00	35	502	0.96
	x12a-0.4	150	150	4.29	150	150	4.29	355	90	-0.40	1.00	35	498	0.95
	x12a-0.6	150	150	4.29	150	150	4.29	355	90	-0.60	1.00	35	484	0.93
	x12a-0.8	150	150	4.29	150	150	4.29	355	90	-0.80	1.00	35	459	0.88
	Kuhn/2019	Para_1	200	80	8.00	200	400	11.00	398	90	0	5.00	10	3239
Para_26		200	80	8.00	200	400	11.00	398	90	-0.25	5.00	10	3252	1.16
Para_51		200	80	8.00	200	400	11.00	398	90	-0.50	5.00	10	3252	1.16
Para_76		200	80	8.00	200	400	11.00	398	90	-0.75	5.00	10	3237	1.15
Para_101		200	80	8.00	200	400	11.00	398	90	0.25	5.00	10	3225	1.15
Para_126		200	80	8.00	200	400	11.00	398	90	0.50	5.00	10	3186	1.14
Para_151		200	80	8.00	200	400	11.00	398	90	0.75	5.00	10	3135	1.12
Para_2		200	160	8.00	200	400	11.00	398	90	0	2.50	20	2641	0.94
Para_27		200	160	8.00	200	400	11.00	398	90	-0.25	2.50	20	2676	0.95
Para_52		200	160	8.00	200	400	11.00	398	90	-0.50	2.50	20	2687	0.96
Para_77		200	160	8.00	200	400	11.00	398	90	-0.75	2.50	20	2673	0.95
Para_102		200	160	8.00	200	400	11.00	398	90	0.25	2.50	20	2608	0.93
Para_127		200	160	8.00	200	400	11.00	398	90	0.50	2.50	20	2530	0.90
Para_152		200	160	8.00	200	400	11.00	398	90	0.75	2.50	20	2392	0.85
Para_6		200	80	8.00	200	200	11.00	398	90	0	2.50	10	1742	1.14
Para_31		200	80	8.00	200	200	11.00	398	90	-0.25	2.50	10	1747	1.14
Para_56		200	80	8.00	200	200	11.00	398	90	-0.50	2.50	10	1747	1.14
Para_106		200	80	8.00	200	200	11.00	398	90	0.25	2.50	10	1737	1.14
Para_131		200	80	8.00	200	200	11.00	398	90	0.50	2.50	10	1720	1.13
Para_156		200	80	8.00	200	200	11.00	398	90	0.75	2.50	10	1694	1.11
P_1		80	80	8.00	80	160	11.00	398	90	0	2.00	10	1423	1.12
P_26		80	80	8.00	80	160	11.00	398	90	-0.25	2.00	10	1434	1.13
P_51		80	80	8.00	80	160	11.00	398	90	-0.50	2.00	10	1433	1.12
P_76		80	80	8.00	80	160	11.00	398	90	-0.75	2.00	10	1422	1.12
P_101		80	80	8.00	80	160	11.00	398	90	0.25	2.00	10	1401	1.10
P_126		80	80	8.00	80	160	11.00	398	90	0.50	2.00	10	1360	1.07
P_151		80	80	8.00	80	160	11.00	398	90	0.75	2.00	10	1292	1.01
P_2		160	160	8.00	160	320	11.00	398	90	0	2.00	20	2201	0.96
P_27		160	160	8.00	160	320	11.00	398	90	-0.25	2.00	20	2228	0.97
P_52		160	160	8.00	160	320	11.00	398	90	-0.50	2.00	20	2237	0.98
P_77		160	160	8.00	160	320	11.00	398	90	-0.75	2.00	20	2237	0.98
P_102		160	160	8.00	160	320	11.00	398	90	0.25	2.00	20	2170	0.95
P_127		160	160	8.00	160	320	11.00	398	90	0.50	2.00	20	2088	0.91
P_152		160	160	8.00	160	320	11.00	398	90	0.75	2.00	20	1969	0.86
P_3		240	240	8.00	240	480	11.00	398	90	0	2.00	30	2504	0.76
P_28	240	240	8.00	240	480	11.00	398	90	-0.25	2.00	30	2549	0.77	
P_53	240	240	8.00	240	480	11.00	398	90	-0.50	2.00	30	2561	0.77	

P_78	240	240	8.00	240	480	11.00	398	90	-0.75	2.00	30	2532	0.76
P_103	240	240	8.00	240	480	11.00	398	90	0.25	2.00	30	2465	0.74
P_128	240	240	8.00	240	480	11.00	398	90	0.50	2.00	30	2408	0.73
P_153	240	240	8.00	240	480	11.00	398	90	0.75	2.00	30	2257	0.68
Para_3	200	240	8.00	200	400	11.00	398	90	0	1.67	30	2198	0.78
Para_28	200	240	8.00	200	400	11.00	398	90	-0.25	1.67	30	2235	0.80
Para_53	200	240	8.00	200	400	11.00	398	90	-0.50	1.67	30	2236	0.80
Para_78	200	240	8.00	200	400	11.00	398	90	-0.75	1.67	30	2217	0.79
Para_103	200	240	8.00	200	400	11.00	398	90	0.25	1.67	30	2166	0.77
Para_128	200	240	8.00	200	400	11.00	398	90	0.50	1.67	30	2112	0.75
Para_153	200	240	8.00	200	400	11.00	398	90	0.75	1.67	30	1967	0.70
Para_7	200	160	8.00	200	200	11.00	398	90	0	1.25	20	1511	0.99
Para_32	200	160	8.00	200	200	11.00	398	90	-0.25	1.25	20	1526	1.00
Para_57	200	160	8.00	200	200	11.00	398	90	-0.50	1.25	20	1516	0.99
Para_82	200	160	8.00	200	200	11.00	398	90	-0.75	1.25	20	1502	0.98
Para_107	200	160	8.00	200	200	11.00	398	90	0.25	1.25	20	1498	0.98
Para_132	200	160	8.00	200	200	11.00	398	90	0.50	1.25	20	1456	0.95
Para_157	200	160	8.00	200	200	11.00	398	90	0.75	1.25	20	1384	0.91
Para_11	200	80	8.00	200	100	11.00	398	90	0	1.25	10	987	1.11
Para_36	200	80	8.00	200	100	11.00	398	90	-0.25	1.25	10	984	1.10
Para_61	200	80	8.00	200	100	11.00	398	90	-0.50	1.25	10	974	1.09
Para_86	200	80	8.00	200	100	11.00	398	90	-0.75	1.25	10	957	1.07
Para_111	200	80	8.00	200	100	11.00	398	90	0.25	1.25	10	990	1.11
Para_136	200	80	8.00	200	100	11.00	398	90	0.50	1.25	10	987	1.11
Para_161	200	80	8.00	200	100	11.00	398	90	0.75	1.25	10	975	1.09
P_6	80	80	8.00	80	80	11.00	398	90	0	1.00	10	823	1.08
P_31	80	80	8.00	80	80	11.00	398	90	-0.25	1.00	10	826	1.08
P_56	80	80	8.00	80	80	11.00	398	90	-0.50	1.00	10	817	1.07
P_81	80	80	8.00	80	80	11.00	398	90	-0.75	1.00	10	795	1.04
P_106	80	80	8.00	80	80	11.00	398	90	0.25	1.00	10	810	1.06
P_131	80	80	8.00	80	80	11.00	398	90	0.50	1.00	10	775	1.01
P_156	80	80	8.00	80	80	11.00	398	90	0.75	1.00	10	706	0.92
P_7	160	160	8.00	160	160	11.00	398	90	0	1.00	20	1279	1.00
P_32	160	160	8.00	160	160	11.00	398	90	-0.25	1.00	20	1287	1.01
P_57	160	160	8.00	160	160	11.00	398	90	-0.50	1.00	20	1274	1.00
P_82	160	160	8.00	160	160	11.00	398	90	-0.75	1.00	20	1250	0.98
P_107	160	160	8.00	160	160	11.00	398	90	0.25	1.00	20	1266	0.99
P_132	160	160	8.00	160	160	11.00	398	90	0.50	1.00	20	1223	0.96
P_157	160	160	8.00	160	160	11.00	398	90	0.75	1.00	20	1145	0.90
P_8	240	240	8.00	240	240	11.00	398	90	0	1.00	30	1547	0.87
P_33	240	240	8.00	240	240	11.00	398	90	-0.25	1.00	30	1547	0.87
P_58	240	240	8.00	240	240	11.00	398	90	-0.50	1.00	30	1530	0.86
P_83	240	240	8.00	240	240	11.00	398	90	-0.75	1.00	30	1490	0.84
P_108	240	240	8.00	240	240	11.00	398	90	0.25	1.00	30	1551	0.87
P_133	240	240	8.00	240	240	11.00	398	90	0.50	1.00	30	1507	0.84
P_158	240	240	8.00	240	240	11.00	398	90	0.75	1.00	30	1418	0.80
Para_8	200	240	8.00	200	200	11.00	398	90	0	0.83	30	1385	0.91
Para_33	200	240	8.00	200	200	11.00	398	90	-0.25	0.83	30	1380	0.90
Para_58	200	240	8.00	200	200	11.00	398	90	-0.50	0.83	30	1352	0.88
Para_83	200	240	8.00	200	200	11.00	398	90	-0.75	0.83	30	1302	0.85
Para_108	200	240	8.00	200	200	11.00	398	90	0.25	0.83	30	1385	0.91
Para_133	200	240	8.00	200	200	11.00	398	90	0.50	0.83	30	1340	0.88
Para_158	200	240	8.00	200	200	11.00	398	90	0.75	0.83	30	1246	0.82
Para_12	200	160	8.00	200	100	11.00	398	90	0	0.63	20	933	1.05
Para_37	200	160	8.00	200	100	11.00	398	90	-0.25	0.63	20	936	1.05
Para_62	200	160	8.00	200	100	11.00	398	90	-0.50	0.63	20	918	1.03
Para_87	200	160	8.00	200	100	11.00	398	90	-0.75	0.63	20	881	0.99
Para_112	200	160	8.00	200	100	11.00	398	90	0.25	0.63	20	930	1.04
Para_137	200	160	8.00	200	100	11.00	398	90	0.50	0.63	20	908	1.02
Para_162	200	160	8.00	200	100	11.00	398	90	0.75	0.63	20	863	0.97
Para_16	200	80	8.00	200	50	11.00	398	90	0	0.63	10	626	1.09
Para_41	200	80	8.00	200	50	11.00	398	90	-0.25	0.63	10	618	1.08
Para_66	200	80	8.00	200	50	11.00	398	90	-0.50	0.63	10	603	1.05

Para_91	200	80	8.00	200	50	11.00	398	90	-0.75	0.63	10	581	1.01
Para_116	200	80	8.00	200	50	11.00	398	90	0.25	0.63	10	632	1.10
Para_141	200	80	8.00	200	50	11.00	398	90	0.50	0.63	10	633	1.10
Para_166	200	80	8.00	200	50	11.00	398	90	0.75	0.63	10	629	1.10
P_11	80	80	8.00	80	40	11.00	398	90	0	0.50	10	596	1.17
P_36	80	80	8.00	80	40	11.00	398	90	-0.25	0.50	10	596	1.17
P_61	80	80	8.00	80	40	11.00	398	90	-0.50	0.50	10	585	1.15
P_86	80	80	8.00	80	40	11.00	398	90	-0.75	0.50	10	558	1.10
P_111	80	80	8.00	80	40	11.00	398	90	0.25	0.50	10	576	1.13
P_136	80	80	8.00	80	40	11.00	398	90	0.50	0.50	10	540	1.06
P_161	80	80	8.00	80	40	11.00	398	90	0.75	0.50	10	476	0.93
P_12	160	160	8.00	160	80	11.00	398	90	0	0.50	20	824	1.08
P_37	160	160	8.00	160	80	11.00	398	90	-0.25	0.50	20	822	1.08
P_62	160	160	8.00	160	80	11.00	398	90	-0.50	0.50	20	800	1.05
P_87	160	160	8.00	160	80	11.00	398	90	-0.75	0.50	20	758	0.99
P_112	160	160	8.00	160	80	11.00	398	90	0.25	0.50	20	820	1.07
P_137	160	160	8.00	160	80	11.00	398	90	0.50	0.50	20	799	1.05
P_162	160	160	8.00	160	80	11.00	398	90	0.75	0.50	20	749	0.98
P_13	240	240	8.00	240	120	11.00	398	90	0	0.50	30	1039	1.02
P_38	240	240	8.00	240	120	11.00	398	90	-0.25	0.50	30	1025	1.01
P_63	240	240	8.00	240	120	11.00	398	90	-0.50	0.50	30	984	0.97
P_88	240	240	8.00	240	120	11.00	398	90	-0.75	0.50	30	914	0.90
P_138	240	240	8.00	240	120	11.00	398	90	0.50	0.50	30	1020	1.00
P_163	240	240	8.00	240	120	11.00	398	90	0.75	0.50	30	964	0.95
Para_13	200	240	8.00	200	100	11.00	398	90	0	0.42	30	949	1.06
Para_38	200	240	8.00	200	100	11.00	398	90	-0.25	0.42	30	936	1.05
Para_63	200	240	8.00	200	100	11.00	398	90	-0.50	0.42	30	892	1.00
Para_88	200	240	8.00	200	100	11.00	398	90	-0.75	0.42	30	816	0.91
Para_113	200	240	8.00	200	100	11.00	398	90	0.25	0.42	30	958	1.07
Para_138	200	240	8.00	200	100	11.00	398	90	0.50	0.42	30	935	1.05
Para_163	200	240	8.00	200	100	11.00	398	90	0.75	0.42	30	878	0.98
Para_17	200	160	8.00	200	50	11.00	398	90	0	0.31	20	651	1.14
Para_42	200	160	8.00	200	50	11.00	398	90	-0.25	0.31	20	637	1.11
Para_67	200	160	8.00	200	50	11.00	398	90	-0.50	0.31	20	610	1.06
Para_92	200	160	8.00	200	50	11.00	398	90	-0.75	0.31	20	569	0.99
Para_117	200	160	8.00	200	50	11.00	398	90	0.25	0.31	20	661	1.15
Para_142	200	160	8.00	200	50	11.00	398	90	0.50	0.31	20	661	1.15
Para_18	200	240	8.00	200	50	11.00	398	90	0	0.21	30	662	1.15
Para_43	200	240	8.00	200	50	11.00	398	90	-0.25	0.21	30	646	1.13
Para_68	200	240	8.00	200	50	11.00	398	90	-0.50	0.21	30	609	1.06
Para_93	200	240	8.00	200	50	11.00	398	90	-0.75	0.21	30	551	0.96
Para_118	200	240	8.00	200	50	11.00	398	90	0.25	0.21	30	687	1.20
Para_143	200	240	8.00	200	50	11.00	398	90	0.50	0.21	30	693	1.21
Para_168	200	240	8.00	200	50	11.00	398	90	0.75	0.21	30	677	1.18
Para_167	200	160	8.00	200	50	11.00	398	90	0.75	0.31	20	643	1.12
P_18	240	240	8.00	240	60	11.00	398	90	0	0.25	30	730	1.15
P_43	240	240	8.00	240	60	11.00	398	90	-0.25	0.25	30	712	1.12
P_68	240	240	8.00	240	60	11.00	398	90	-0.50	0.25	30	676	1.06
P_93	240	240	8.00	240	60	11.00	398	90	-0.75	0.25	30	613	0.96
P_118	240	240	8.00	240	60	11.00	398	90	0.25	0.25	30	746	1.17
P_143	240	240	8.00	240	60	11.00	398	90	0.50	0.25	30	753	1.18
P_168	240	240	8.00	240	60	11.00	398	90	0.75	0.25	30	743	1.17

Table 10Numerical results of 22 RHS-to-RHS X joints selected to examine the effect of η^* ratio.

Specimen	f_{y0} (MPa)	θ_1 (°)	n	η^*	N_{1u} (kN)	$N_{1u}/N_{C,M}$	N_{1u}/N_{Yu}	N_{1u}/N_{Kuhn}	N_{1u}/N_{Lan}	$N_{1u}/N_{P,M}$	$N_{1u}/N_{P,LK}$
x10ae	355	90	0	1.00	1019	1.11	1.09	1.11	1.06	1.11	1.13
x10ae2	355	90	0	2.00	1507	1.06	1.04	1.05	1.01	1.17	1.19
x11ae05	355	90	0	0.50	2498	1.28	1.15	1.29	1.08	1.15	1.17
x11a	355	90	0	1.00	509	1.14	1.07	1.15	0.96	1.14	1.16
x11ae2	355	90	0	2.00	776	1.15	1.11	1.16	1.06	1.28	1.29
x12ae05	355	90	0	0.50	1427	1.48	1.29	1.49	1.10	1.33	1.32
x12ae	355	90	0	1.00	302	1.33	1.22	1.33	1.11	1.33	1.32
x12ae2	355	90	0	2.00	482	1.20	1.15	1.20	1.19	1.33	1.32
Para_2	398	90	0	2.50	1507	1.08	1.04	1.09	1.01	1.24	1.26
P_2	398	90	0	2.00	2498	1.10	1.07	1.11	1.00	1.22	1.24
P_3	398	90	0	2.00	509	1.03	0.99	1.04	1.00	1.14	1.15
Para_3	398	90	0	1.67	776	1.07	1.03	1.08	1.00	1.16	1.16
Para_7	398	90	0	1.25	1427	1.14	1.06	1.15	1.00	1.18	1.20
P_6	398	90	0	1.00	302	1.09	1.13	1.09	1.09	1.09	1.10
P_7	398	90	0	1.00	482	1.15	1.10	1.16	1.02	1.15	1.17
P_8	398	90	0	1.00	816	1.18	1.10	1.20	1.00	1.18	1.19
Para_8	398	90	0	0.83	2641	1.24	1.16	1.25	1.01	1.20	1.21
Para_12	398	90	0	0.63	2201	1.20	1.08	1.21	1.06	1.12	1.14
Para_16	398	90	0	0.63	2504	1.11	0.93	1.11	1.11	1.11	1.11
P_12	398	90	0	0.50	2198	1.24	1.14	1.25	1.09	1.12	1.14
P_13	398	90	0	0.50	1511	1.39	1.22	1.41	1.06	1.25	1.26
Para_13	398	90	0	0.42	823	1.45	1.30	1.47	1.08	1.27	1.28
Mean							1.19	1.11	1.20	1.05	1.20
CoV							0.105	0.080	0.106	0.051	0.064

Note: The material factor (C_f) was used for all the design methods.

Table 11

Numerical results of 10 RHS-to-RHS X joints selected to examine the chord stress effect.

Specimen	f_{y0} (MPa)	θ_1 (°)	n	η^*	N_{1u} (kN)	$N_{1u}/N_{C,M}$	N_{1u}/N_{Yu}	N_{1u}/N_{Kuhn}	N_{1u}/N_{Lan}	$N_{1u}/N_{P,M}$	$N_{1u}/N_{P,LK}$
x10a-0.6	355	90	-0.6	1.00	1562	1.27	1.24	1.26	1.21	1.27	1.28
x10a-0.8	355	90	-0.8	1.00	1492	1.30	1.27	1.29	1.23	1.30	1.31
x11a-0	355	90	0	1.00	817	1.20	1.13	1.21	1.02	1.20	1.22
x11a-0.4	355	90	-0.4	1.00	803	1.24	1.17	1.26	1.05	1.24	1.26
x11a-0.6	355	90	-0.6	1.00	787	1.27	1.19	1.28	1.07	1.27	1.29
x11a-0.8	355	90	-0.8	1.00	757	1.31	1.23	1.32	1.11	1.31	1.33
x12a-0	355	90	0	1.00	502	1.38	1.27	1.39	1.16	1.38	1.37
x12a-0.4	355	90	-0.4	1.00	498	1.44	1.33	1.45	1.21	1.44	1.43
x12a-0.6	355	90	-0.6	1.00	484	1.46	1.35	1.47	1.22	1.46	1.45
x12a-0.8	355	90	-0.8	1.00	459	1.48	1.37	1.49	1.24	1.48	1.47
Mean						1.34	1.25	1.34	1.15	1.34	1.34
CoV						0.074	0.063	0.073	0.073	0.074	0.064

Note: The material factor (C_T) equals 1.0 for all the joints.

Table 12Evaluation of design methods, using the C_f factor, against numerical results of 131 screened RHS-to-RHS X joints under brace axial compression.

Specimen	f_{y0} (MPa)	θ_1 (°)	n	η^*	N_{1u} (kN)	$N_{1u}/N_{C,M}$	N_{1u}/N_{Y_u}	N_{1u}/N_{Kuhn}	N_{1u}/N_{Lan}	$N_{1u}/N_{P,M}$	$N_{1u}/N_{P,LK}$
x10ae	355	90	0	1.00	1507	1.11	1.09	1.11	1.06	1.11	1.13
x10ae2	355	90	0	2.00	2498	1.06	1.04	1.05	1.01	1.17	1.19
x11ae05	355	90	0	0.50	509	1.28	1.15	1.29	1.08	1.15	1.17
x11a	355	90	0	1.00	776	1.14	1.07	1.15	0.96	1.14	1.16
x11ae2	355	90	0	2.00	1427	1.15	1.11	1.16	1.06	1.28	1.29
x12ae05	355	90	0	0.50	302	1.48	1.29	1.49	1.10	1.33	1.32
x12ae	355	90	0	1.00	482	1.33	1.22	1.33	1.11	1.33	1.32
x12ae2	355	90	0	2.00	816	1.20	1.15	1.20	1.19	1.33	1.32
x10a-0.6	355	90	-0.60	1.00	1562	1.27	1.24	1.26	1.21	1.27	1.28
x10a-0.8	355	90	-0.80	1.00	1492	1.30	1.27	1.29	1.23	1.30	1.31
x11a-0	355	90	0	1.00	817	1.20	1.13	1.21	1.02	1.20	1.22
x11a-0.4	355	90	-0.40	1.00	803	1.24	1.17	1.26	1.05	1.24	1.26
x11a-0.6	355	90	-0.60	1.00	787	1.27	1.19	1.28	1.07	1.27	1.29
x11a-0.8	355	90	-0.80	1.00	757	1.31	1.23	1.32	1.11	1.31	1.33
x12a-0	355	90	0	1.00	502	1.38	1.27	1.39	1.16	1.38	1.37
x12a-0.4	355	90	-0.40	1.00	498	1.44	1.33	1.45	1.21	1.44	1.43
x12a-0.6	355	90	-0.60	1.00	484	1.46	1.35	1.47	1.22	1.46	1.45
x12a-0.8	355	90	-0.80	1.00	459	1.48	1.37	1.49	1.24	1.48	1.47
Para_2	398	90	0	2.50	2641	1.08	1.04	1.09	1.01	1.24	1.26
Para_27	398	90	-0.25	2.50	2676	1.13	1.09	1.14	1.05	1.30	1.32
Para_52	398	90	-0.50	2.50	2687	1.18	1.14	1.19	1.10	1.36	1.38
Para_77	398	90	-0.75	2.50	2673	1.26	1.21	1.27	1.17	1.45	1.47
Para_102	398	90	0.25	2.50	2608	1.10	1.06	1.11	1.03	1.26	1.28
Para_127	398	90	0.50	2.50	2530	1.11	1.07	1.12	1.04	1.28	1.30
Para_152	398	90	0.75	2.50	2392	1.13	1.09	1.14	1.05	1.29	1.31
P_101	398	90	0.25	2.00	1401	1.15	1.17	1.15	1.15	1.27	1.28
P_126	398	90	0.50	2.00	1360	1.16	1.18	1.16	1.16	1.29	1.29
P_151	398	90	0.75	2.00	1292	1.18	1.21	1.18	1.18	1.31	1.32
P_2	398	90	0	2.00	2201	1.10	1.07	1.11	1.00	1.22	1.24
P_27	398	90	-0.25	2.00	2228	1.15	1.12	1.16	1.04	1.28	1.30
P_52	398	90	-0.50	2.00	2237	1.20	1.17	1.21	1.09	1.33	1.36
P_77	398	90	-0.75	2.00	2237	1.29	1.25	1.30	1.17	1.43	1.45
P_102	398	90	0.25	2.00	2170	1.12	1.09	1.13	1.01	1.24	1.26
P_127	398	90	0.50	2.00	2088	1.12	1.09	1.13	1.02	1.25	1.27
P_152	398	90	0.75	2.00	1969	1.13	1.10	1.14	1.03	1.26	1.28
P_3	398	90	0	2.00	2504	1.03	0.99	1.04	1.00	1.14	1.15
P_28	398	90	-0.25	2.00	2549	1.08	1.04	1.09	1.05	1.20	1.20
P_53	398	90	-0.50	2.00	2561	1.13	1.08	1.14	1.10	1.25	1.26
P_78	398	90	-0.75	2.00	2532	1.20	1.15	1.21	1.16	1.33	1.33
P_103	398	90	0.25	2.00	2465	1.04	1.00	1.06	1.01	1.16	1.16
P_128	398	90	0.50	2.00	2408	1.06	1.02	1.07	1.03	1.18	1.18
P_153	398	90	0.75	2.00	2257	1.07	1.02	1.08	1.04	1.18	1.19
Para_3	398	90	0	1.67	2198	1.07	1.03	1.08	1.00	1.16	1.16
Para_28	398	90	-0.25	1.67	2235	1.12	1.08	1.13	1.04	1.21	1.21
Para_53	398	90	-0.50	1.67	2236	1.17	1.12	1.18	1.09	1.26	1.26
Para_78	398	90	-0.75	1.67	2217	1.24	1.19	1.25	1.15	1.34	1.34
Para_103	398	90	0.25	1.67	2166	1.09	1.05	1.10	1.01	1.17	1.18
Para_128	398	90	0.50	1.67	2112	1.10	1.06	1.11	1.02	1.19	1.19
Para_153	398	90	0.75	1.67	1967	1.10	1.06	1.11	1.02	1.19	1.19
Para_7	398	90	0	1.25	1511	1.14	1.06	1.15	1.00	1.18	1.20
Para_32	398	90	-0.25	1.25	1526	1.18	1.10	1.19	1.04	1.22	1.24
Para_57	398	90	-0.50	1.25	1516	1.22	1.14	1.23	1.08	1.26	1.29
Para_82	398	90	-0.75	1.25	1502	1.30	1.21	1.31	1.14	1.34	1.36
Para_107	398	90	0.25	1.25	1498	1.16	1.09	1.17	1.02	1.20	1.22
Para_132	398	90	0.50	1.25	1456	1.17	1.10	1.18	1.03	1.21	1.23
Para_157	398	90	0.75	1.25	1384	1.20	1.12	1.21	1.05	1.24	1.26
Para_36	398	90	-0.25	1.25	984	1.15	1.03	1.15	1.15	1.19	1.20
Para_61	398	90	-0.50	1.25	974	1.19	1.06	1.19	1.19	1.23	1.23

Para_86	398	90	-0.75	1.25	957	1.25	1.12	1.25	1.25	1.29	1.30
Para_161	398	90	0.75	1.25	975	1.27	1.14	1.27	1.27	1.31	1.32
P_6	398	90	0	1.00	823	1.09	1.13	1.09	1.09	1.09	1.10
P_31	398	90	-0.25	1.00	826	1.13	1.17	1.13	1.13	1.13	1.13
P_56	398	90	-0.50	1.00	817	1.16	1.20	1.16	1.16	1.16	1.17
P_81	398	90	-0.75	1.00	795	1.21	1.25	1.21	1.21	1.21	1.22
P_106	398	90	0.25	1.00	810	1.10	1.14	1.10	1.10	1.10	1.11
P_131	398	90	0.50	1.00	775	1.10	1.14	1.10	1.10	1.10	1.11
P_156	398	90	0.75	1.00	706	1.07	1.11	1.07	1.07	1.07	1.08
P_7	398	90	0	1.00	1279	1.15	1.10	1.16	1.02	1.15	1.17
P_32	398	90	-0.25	1.00	1287	1.20	1.14	1.21	1.05	1.20	1.22
P_57	398	90	-0.50	1.00	1274	1.23	1.17	1.24	1.09	1.23	1.25
P_82	398	90	-0.75	1.00	1250	1.30	1.23	1.31	1.14	1.30	1.32
P_107	398	90	0.25	1.00	1266	1.18	1.12	1.19	1.03	1.18	1.20
P_132	398	90	0.50	1.00	1223	1.18	1.12	1.19	1.04	1.18	1.20
P_157	398	90	0.75	1.00	1145	1.19	1.13	1.20	1.04	1.19	1.21
P_8	398	90	0	1.00	1547	1.18	1.10	1.20	1.00	1.18	1.19
P_33	398	90	-0.25	1.00	1547	1.22	1.13	1.23	1.03	1.22	1.22
P_58	398	90	-0.50	1.00	1530	1.25	1.16	1.27	1.06	1.25	1.26
P_83	398	90	-0.75	1.00	1490	1.31	1.21	1.32	1.10	1.31	1.31
P_108	398	90	0.25	1.00	1551	1.22	1.13	1.23	1.03	1.22	1.23
P_133	398	90	0.50	1.00	1507	1.24	1.15	1.25	1.04	1.24	1.24
P_158	398	90	0.75	1.00	1418	1.25	1.16	1.26	1.05	1.25	1.25
Para_8	398	90	0	0.83	1385	1.24	1.16	1.25	1.01	1.20	1.21
Para_33	398	90	-0.25	0.83	1380	1.27	1.19	1.28	1.03	1.23	1.24
Para_58	398	90	-0.50	0.83	1352	1.29	1.21	1.31	1.06	1.26	1.26
Para_83	398	90	-0.75	0.83	1302	1.33	1.25	1.35	1.09	1.30	1.30
Para_108	398	90	0.25	0.83	1385	1.27	1.19	1.29	1.04	1.24	1.24
Para_133	398	90	0.50	0.83	1340	1.28	1.20	1.30	1.05	1.25	1.25
Para_158	398	90	0.75	0.83	1246	1.28	1.19	1.29	1.04	1.24	1.25
Para_12	398	90	0	0.63	933	1.20	1.08	1.21	1.06	1.12	1.14
Para_37	398	90	-0.25	0.63	936	1.24	1.11	1.25	1.09	1.16	1.18
Para_62	398	90	-0.50	0.63	918	1.27	1.13	1.28	1.12	1.18	1.20
Para_87	398	90	-0.75	0.63	881	1.30	1.17	1.32	1.15	1.22	1.24
Para_112	398	90	0.25	0.63	930	1.23	1.10	1.25	1.09	1.15	1.17
Para_137	398	90	0.50	0.63	908	1.25	1.12	1.27	1.10	1.17	1.19
Para_162	398	90	0.75	0.63	863	1.28	1.14	1.29	1.12	1.19	1.21
Para_16	398	90	0	0.63	626	1.11	0.93	1.11	1.11	1.11	1.11
Para_41	398	90	-0.25	0.63	618	1.12	0.95	1.12	1.12	1.12	1.12
Para_66	398	90	-0.50	0.63	603	1.14	0.96	1.14	1.14	1.14	1.14
Para_91	398	90	-0.75	0.63	581	1.18	0.99	1.18	1.18	1.18	1.18
Para_116	398	90	0.25	0.63	632	1.15	0.97	1.15	1.15	1.15	1.15
Para_141	398	90	0.50	0.63	633	1.20	1.01	1.20	1.20	1.20	1.20
Para_166	398	90	0.75	0.63	629	1.27	1.08	1.27	1.27	1.27	1.27
P_86	398	90	-0.75	0.50	558	1.27	1.35	1.27	1.27	1.27	1.27
P_136	398	90	0.50	0.50	540	1.15	1.21	1.15	1.15	1.15	1.15
P_161	398	90	0.75	0.50	476	1.09	1.15	1.09	1.09	1.09	1.09
P_12	398	90	0	0.50	824	1.24	1.14	1.25	1.09	1.12	1.14
P_37	398	90	-0.25	0.50	822	1.27	1.17	1.28	1.12	1.15	1.17
P_62	398	90	-0.50	0.50	800	1.29	1.19	1.30	1.14	1.16	1.18
P_87	398	90	-0.75	0.50	758	1.31	1.20	1.32	1.15	1.18	1.20
P_112	398	90	0.25	0.50	820	1.27	1.17	1.28	1.12	1.14	1.16
P_137	398	90	0.50	0.50	799	1.29	1.18	1.30	1.13	1.16	1.18
P_162	398	90	0.75	0.50	749	1.29	1.19	1.30	1.14	1.17	1.19
P_13	398	90	0	0.50	1039	1.39	1.22	1.41	1.06	1.25	1.26
P_38	398	90	-0.25	0.50	1025	1.41	1.24	1.43	1.07	1.27	1.28
P_63	398	90	-0.50	0.50	984	1.41	1.24	1.43	1.07	1.27	1.28
P_88	398	90	-0.75	0.50	914	1.41	1.24	1.42	1.07	1.27	1.27
P_138	398	90	0.50	0.50	1020	1.46	1.29	1.48	1.11	1.32	1.33
P_163	398	90	0.75	0.50	964	1.48	1.30	1.50	1.13	1.34	1.34
Para_13	398	90	0	0.42	949	1.45	1.30	1.47	1.08	1.27	1.28
Para_38	398	90	-0.25	0.42	936	1.47	1.32	1.49	1.09	1.29	1.30
Para_63	398	90	-0.50	0.42	892	1.46	1.31	1.48	1.09	1.28	1.29

Para_88	398	90	-0.75	0.42	816	1.43	1.28	1.45	1.06	1.26	1.26
Para_113	398	90	0.25	0.42	958	1.51	1.35	1.52	1.12	1.32	1.33
Para_138	398	90	0.50	0.42	935	1.53	1.37	1.55	1.14	1.34	1.35
Para_163	398	90	0.75	0.42	878	1.54	1.38	1.56	1.15	1.35	1.36
Para_67	398	90	-0.50	0.31	610	1.31	1.11	1.32	1.15	1.15	1.15
Para_92	398	90	-0.75	0.31	569	1.31	1.11	1.32	1.15	1.15	1.15
Para_68	398	90	-0.50	0.21	609	1.55	1.31	1.57	1.15	1.23	1.23
Para_93	398	90	-0.75	0.21	551	1.51	1.27	1.52	1.12	1.19	1.20
P_68	398	90	-0.50	0.25	676	1.55	1.27	1.57	1.15	1.26	1.27
P_93	398	90	-0.75	0.25	613	1.51	1.24	1.52	1.12	1.23	1.23
Mean						1.24	1.15	1.25	1.10	1.23	1.24
CoV						0.102	0.082	0.104	0.061	0.065	0.064

Note: The material factor (C_f) was used for all the design methods.

Table 13

Collated test results totalling 24 RHS X joints with only one RHS brace welded to the chord and under brace axial compression.

Researcher/year	Specimen	b_0 (mm)	h_0 (mm)	t_0 (mm)	b_1 (mm)	h_1 (mm)	t_1 (mm)	f_{y0} (MPa)	θ_1 (°)	n	$2\gamma^*$	η^*	N_{1u} (kN)	N_{1u}/N_y
Barentse/1977	T-RR-A-A-1	101.4	101.4	6.23	101.4	70.0	6.23	299	90	0	16.3	0.69	417	1.11
	T-RR-A-A-7	101.3	101.3	4.03	100.2	70.0	3.77	326	90	0	25.1	0.69	210	0.89
	T-RR-A-A-10	100.4	100.4	2.88	100.4	70.0	2.88	299	90	0	34.9	0.70	112	0.77
	T-RR-E-A-91	101.3	50.9	6.27	101.3	50.9	6.27	322	90	0	8.1	1.00	453	1.36
	T-RR-E-A-94	101.8	50.9	4.73	101.6	50.8	4.90	338	90	0	10.8	1.00	298	1.25
	T-RR-E-A-97	101.3	51.0	3.35	101.3	51.0	3.35	293	90	0	15.2	1.00	150	1.13
	T-RR-E-A-112	102.1	151.3	6.23	101.4	70.0	6.18	289	90	0	24.3	0.46	394	1.08
	T-RR-E-A-119	101.2	151.4	4.82	101.4	70.0	6.23	293	90	0	31.4	0.46	256	0.96
	T-RR-E-A-126	80.6	119.7	3.05	80.2	70.0	3.00	398	90	0	39.2	0.58	142	0.69
	T-RR-C-A-46	100.2	100.2	3.77	100.0	25.0	3.08	349	90	0	26.6	0.25	180	1.16
	T-RR-C-A-50	100.8	100.8	3.36	100.0	25.0	3.02	290	90	0	30.0	0.25	152	1.37
	T-RR-C-A-41	101.4	101.4	6.18	100.9	70.0	3.92	298	90	0	16.4	0.69	394	1.06
	T-RR-C-A-45	101.0	101.0	4.00	100.9	70.0	3.92	343	90	0	25.3	0.69	248	1.00
	T-RR-C-A-49	100.8	100.8	3.36	99.8	70.0	4.00	290	90	0	30.0	0.69	163	0.96
Zhao/2000	S1B1C11	51.0	102.0	4.90	51.0	51.0	4.90	409	90	0	20.8	0.50	316	1.04
	S1B1C12	51.0	102.0	3.20	51.0	51.0	4.90	343	90	0	31.9	0.50	163	1.11
	S1B2C21	102.0	102.0	9.50	102.0	102.0	8.00	445	90	0	10.7	1.00	1207	0.95
	S1B2C22	102.0	102.0	6.30	102.0	102.0	8.00	432	90	0	16.2	1.00	652	0.90
Pandey/2019	TF-100x50x4- 100x50x4	100.6	50.6	3.96	100.6	50.6	3.97	952	90	0	12.8	1.00	494	0.93
	TF-120x120x4- 120x120x4	121.6	121.6	3.91	121.6	121.7	3.91	971	90	0	31.1	1.00	558	0.52
	TF-140x140x4- 140x140x4	141.6	140.3	3.97	141.7	140.4	4.00	1008	90	0	35.3	1.00	544	0.42
	TF-120x120x3- 120x120x3	120.9	120.3	3.12	120.8	120.3	3.11	1038	90	0	38.5	1.00	369	0.42
Fan/2017	X-1.0-32-700O	203.6	203.6	5.96	204.0	204.0	11.67	404	90	0	34.2	1.00	653	0.58
	X-1.0-21-550O	203.1	203.1	8.85	204.0	204.0	11.67	418	90	0	22.9	1.00	1264	0.69

Table 14Evaluation of proposed design methods, using the C_f factor, against test results of eight RHS X joints classified as class a.

Specimen	f_{y0} (MPa)	θ_1 (°)	n	η^*	N_{1u} (kN)	$N_{1u}/N_{P,M}$	$N_{1u}/N_{P,LK}$
T-RR-A-A-7	326	90	0	0.69	210	1.00	1.01
T-RR-A-A-10	299	90	0	0.70	112	0.99	0.99
T-RR-E-A-112	289	90	0	0.46	394	1.11	1.12
T-RR-E-A-119	293	90	0	0.46	256	1.10	1.10
T-RR-E-A-126	398	90	0	0.58	142	1.06	1.03
T-RR-C-A-41	298	90	0	0.69	394	1.06	1.07
T-RR-C-A-45	343	90	0	0.69	248	1.14	1.16
T-RR-C-A-49	290	90	0	0.69	163	1.14	1.15
Mean						1.07	1.08
CoV						0.054	0.059

Table 15Evaluation of proposed design methods, using the C_f factor, against test results of six RHS X joints classified as class b.

Specimen	f_{y0} (MPa)	θ_1 (°)	n	$2\gamma^*$	$2\gamma^*$ limit	η^*	N_{1u} (kN)	$N_{1u}/N_{P,M}$	$N_{1u}/N_{P,LK}$
S1B1C11	409	90	0	20.8	37	0.50	316	1.26	1.29
S1B2C21	445	90	0	10.7	36	1.00	1207	1.04	1.07
S1B2C22	432	90	0	16.2	36	1.00	652	1.09	1.12
TF-100x50x4-100x50x4	952	90	0	12.8	24	1.00	494	1.39	1.47
X-1.0-32-700O	404	90	0	34.2	37	1.00	653	1.20	1.18
X-1.0-21-550O	418	90	0	22.9	37	1.00	1264	0.98	1.01
Mean								1.16	1.19
CoV								0.130	0.141

Table 16

Collated numerical results totalling eight RHS-to-RHS X joints under brace in-plane bending, reported by Yu [19].

Specimen	b_0 (mm)	h_0 (mm)	t_0 (mm)	b_1 (mm)	h_1 (mm)	t_1 (mm)	f_{y0} (MPa)	θ_1 (°)	n	$2\gamma^*$	η^*	$M_{1u,ip}$ (kNm)	$M_{1u,ip}/M_{y,ip}$
x10ie05	150	150	10.00	150	75	10.00	355	90	0	15	0.5	37.1	1.34
x10ie	150	150	10.00	150	150	10.00	355	90	0	15	1.0	89.7	1.26
x10ie2	150	150	10.00	150	300	10.00	355	90	0	15	2.0	259.7	1.19
x11i	150	150	6.25	150	150	6.25	355	90	0	24	1.0	50.0	1.37
x11ie2	150	150	6.25	150	300	6.25	355	90	0	24	2.0	128.7	1.06
x12ie05	150	150	4.29	150	75	4.29	355	90	0	35	0.5	12.2	1.72
x12i	150	150	4.29	150	150	4.29	355	90	0	35	1.0	28.6	1.28
x12ie2	150	150	4.29	150	300	4.29	355	90	0	35	2.0	76.5	0.97

Table 17

Evaluation of design methods for the loading case of brace in-plane bending.

Specimen	f_{y0} (MPa)	$2\gamma^*$	η^*	$M_{1u,ip}$ (kNm)	$M_{1u,ip}/M_{C,M,ip}$	$M_{1u,ip}/M_{Yu,ip}$	$M_{1u,ip}/M_{Kuhn,ip}$	$M_{1u,ip}/M_{Lan,ip}$	$M_{1u,ip}/M_{P,M,ip}$	$M_{1u,ip}/M_{P,LK,ip}$
x10ie05	355	15	0.5	37.1	1.40	1.36	1.40	1.34	1.34	1.34
x10ie	355	15	1.0	89.7	1.33	1.25	1.32	1.26	1.33	1.34
x10ie2	355	15	2.0	259.7	1.25	1.24	1.25	1.19	1.39	1.41
x11i	355	24	1.0	50.0	1.63	1.24	1.64	1.37	1.63	1.65
x11ie2	355	24	2.0	128.7	1.25	1.18	1.26	1.15	1.39	1.41
x12ie05	355	35	0.5	12.2	2.48	1.38	2.48	1.84	2.23	2.22
x12i	355	35	1.0	28.6	1.84	1.11	1.84	1.54	1.84	1.82
x12ie2	355	35	2.0	76.5	1.40	1.29	1.40	1.40	1.55	1.54
Mean					1.57	1.25	1.57	1.39	1.59	1.59
CoV					0.265	0.072	0.265	0.159	0.198	0.191

Note: The material factor (C_T) equals 1.0 for all the joints.

Table 18

Effects of chord sidewall convexity and thickness tolerance on the resistance of RHS-to-RHS T joints under brace in-plane bending (Nagui [42])

Specimen	2γ	Steel grade	$M_{1u,ip}$ (kNm)	$M_{C,ip}$ (kNm)	$M_{1u,ip}/M_{C,ip}$
t12i (convexity: 1%)	35	S355	28.9	22.9	1.26
	35	S460	34.5	29.7	1.16
	35	S700	50.3	46.1	1.09
t12i* (convexity: 1% + thickness : -10%)	35	S355	24.7	22.9	1.08
	35	S460	29.7	29.7	1.00
	35	S700	43.2	46.1	0.94
t13i (convexity: 1%)	30	S355	37.2	27.2	1.37
	30	S460	44.7	36.4	1.22
	30	S700	64.8	55.4	1.17
t13i* (convexity: 1% + thickness : -10%)	30	S355	31.9	27.2	1.17
	30	S460	38.4	36.4	1.05
	30	S700	55.8	55.4	1.01
Mean					1.13
CoV					0.109

Table 19

Collated numerical results totalling eight RHS-to-RHS X joints under brace out-of-plane bending, reported by Yu [19].

Specimen	b_0 (mm)	h_0 (mm)	t_0 (mm)	b_1 (mm)	h_1 (mm)	t_1 (mm)	f_{y0} (MPa)	θ_1 (°)	n	$2\gamma^*$	η^*	$M_{1u,op}$ (kNm)	$M_{1u,op}/M_{y,op}$
x10oe05	150	150	10.00	150	75	10.00	355	90	0	15	0.5	80.4	1.29
x10oe	150	150	10.00	150	150	10.00	355	90	0	15	1.0	119.4	1.20
x10oe2	150	150	10.00	150	300	10.00	355	90	0	15	2.0	192.5	1.11
x11o	150	150	6.25	150	150	6.25	355	90	0	24	1.0	59.4	1.03
x11oe2	150	150	6.25	150	300	6.25	355	90	0	24	2.0	108.6	1.03
x12oe05	150	150	4.29	150	75	4.29	355	90	0	35	0.5	23.0	1.08
x12o	150	150	4.29	150	150	4.29	355	90	0	35	1.0	37.2	0.98
x12oe2	150	150	4.29	150	300	4.29	355	90	0	35	2.0	62.7	0.88

Table 20

Evaluation of design methods for the loading case of brace out-of-plane bending.

Specimen	f_{y0} (MPa)	$2\gamma^*$	η^*	$M_{1u,op}$ (kNm)	$M_{1u,op}/M_{C,M,op}$	$M_{1u,op}/M_{Yu,op}$	$M_{1u,op}/M_{Kuhn,op}$	$M_{1u,op}/M_{Lan,op}$	$M_{1u,op}/M_{P,M,op}$	$M_{1u,op}/M_{P,LK,op}$
x10oe05	355	15	0.5	80.4	1.36	1.21	1.36	1.29	1.29	1.29
x10oe	355	15	1.0	119.4	1.26	1.14	1.26	1.20	1.26	1.28
x10oe2	355	15	2.0	192.5	1.16	1.06	1.16	1.11	1.29	1.31
x11o	355	24	1.0	59.4	1.22	1.09	1.23	1.03	1.22	1.23
x11oe2	355	24	2.0	108.6	1.22	1.12	1.23	1.12	1.35	1.37
x12oe05	355	35	0.5	23.0	1.55	1.30	1.56	1.15	1.39	1.39
x12o	355	35	1.0	37.2	1.41	1.26	1.41	1.18	1.41	1.40
x12oe2	355	35	2.0	62.7	1.26	1.17	1.27	1.25	1.40	1.39
Mean					1.30	1.17	1.31	1.17	1.33	1.33
CoV					0.097	0.070	0.097	0.073	0.054	0.046

Note: The material factor (C_T) equals 1.0 for all the joints.

Table 21

Recommended design resistance for chord sidewall failure in RHS joints under brace axial compression using the modified bearing-buckling method.

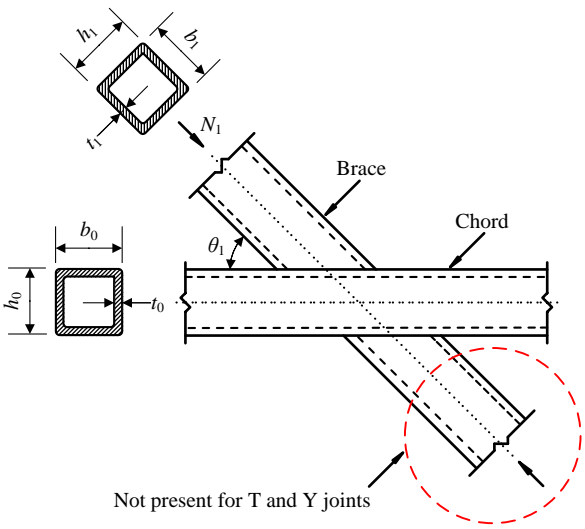
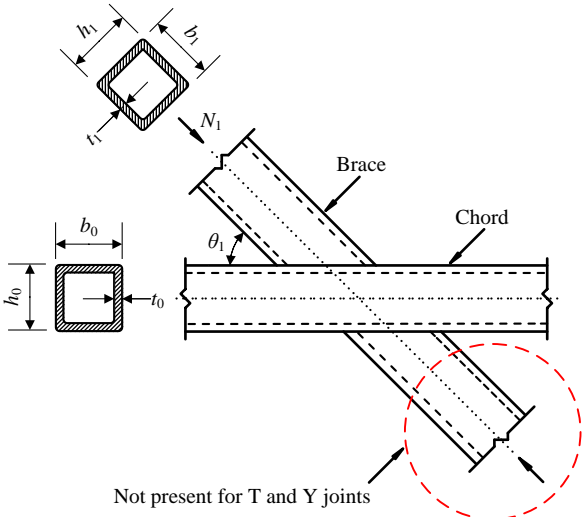
RHS-to-RHS X, T and Y joints (class a)	Brace axial compression loading	
 <p>Not present for T and Y joints</p>	$N_{1,Rd} = C_f f_k t_0 (2h_1 + 10t_0) \sqrt{\frac{1}{\sin \theta_1}} Q_f$	$f_k = \chi_{0.5} \left(\frac{h_0}{h_1} \right)^{0.15} \quad f_{y0} \leq f_{y0}$
	Parameters	
	$C_f = 1.1 - 0.1 f_{y0} / 355 \leq 1.0$	
	$\chi_{0.5}$ is the reduction factor for column buckling according to e.g., EN 1993-1-1 [6] using the buckling curve c, or an equivalent code/standard, and a normalised slenderness: $\lambda_{0.5} = \frac{1.73 \left(\frac{h_0}{t_0} - 2 \right)}{\pi \sqrt{\frac{E}{f_{y0}}}}$	
	$Q_f = (1 - n)^{0.1}$ with n in the connecting chord face $n = \frac{N_{0,Ed}}{N_{pl,0,Rd}} + \frac{M_{0,Ed}}{M_{pl,0,Rd}}$ for class 1 or 2 chord cross-sections under chord compression stress and for chord cross-sections under chord tension stress; $M_{el,Rd}$ should be used for class 3 chord cross-sections.	
Validity ranges		
steel grades up to S960; $\beta=1.0$; $b_0/t_0 \leq 40$, $h_0/t_0 \leq 40$; $0.25 \leq h_1/h_0 \leq 2.0$; $0.5 \leq h_0/b_0 \leq 2.0$; $\theta_1 \geq 30^\circ$.		
<p>These recommendations may also be used for other X joints under brace axial compression:</p> <ol style="list-style-type: none"> (1) Plate-to-RHS X joints (use $t_1=h_1$) and RHS-to-RHS X joints both with nominal $f_{y0} \leq 460$ MPa, $\beta=1.0$ and $h_1/h_0 \leq 0.25$, but use $f_k=f_{y0}$. (2) For X joints (class a) with an RHS brace welded to one side of the chord and supported by a flat plate or another profile welded to the opposite side of the chord, the lower effective h_1 on either side and $\lambda_{0.5}$ should be used for the determination of the resistance. (3) For X joints (class b) with an RHS brace welded to one side of the chord and supported by a flat plate or another profile unwelded to the opposite side of the chord, the lower effective h_1 on either side and $\lambda_{0.7}=1.4\lambda_{0.5}$ should be used for the determination of the resistance. Here, the chord cross-section slenderness should not exceed class 3. (4) For other unspecified X joints (class c), the resistance should be determined using $\lambda_{1.0}=2\lambda_{0.5}$. 		
<p>The cross-section slenderness of class 1, 2 and 3 is defined in national standards.</p>		

Table 22
Recommended design resistance for chord sidewall failure in RHS joints under brace axial compression using the Lan-Kuhn method.

RHS-to-RHS X, T and Y joints (class a)	Brace axial compression loading	
 <p>Not present for T and Y joints</p>	$N_{1,Rd} = C_f f_k t_0 (2h_1 + 10t_0) \sqrt{\frac{1}{\sin \theta_1}} Q_f$	$f_k = \chi_{0.5} \left(\frac{h_0}{h_1} \right)^{0.15} \quad f_{y0} \leq f_{y0}$
	Parameters	
	$C_f = 1.1 - 0.1 f_{y0} / 355 \leq 1.0$	
	$\chi_{0.5}$ is the buckling reduction factor obtained from: $\chi_{0.5} = 1.12 - 0.012 \frac{h_0}{t_0} \sqrt{\frac{f_{y0}}{355}}$	
	$Q_f = (1 - n)^{0.1}$ with n in connecting chord face $n = \frac{N_{0,Ed}}{N_{pl,0,Rd}} + \frac{M_{0,Ed}}{M_{pl,0,Rd}}$ for class 1 or 2 chord cross-sections under chord compression stress and for chord cross-sections under chord tension stress; $M_{el,Rd}$ should be used for class 3 chord cross-sections.	
Validity ranges		
steel grades up to S960; $\beta=1.0$; $b_0/t_0 \leq 40$, $h_0/t_0 \leq 40$; $0.25 \leq h_1/h_0 \leq 2.0$; $0.5 \leq h_0/b_0 \leq 2.0$; $\theta_1 \geq 30^\circ$.		
These recommendations may also be used for other X joints under brace axial compression:		
<ol style="list-style-type: none"> (1) Plate-to-RHS X joints (use $t_1=h_1$) and RHS-to-RHS X joints both with nominal $f_{y0} \leq 460$ MPa, $\beta=1.0$ and $h_1/h_0 \leq 0.25$, but use $f_k=f_{y0}$. (2) For X joints (class a) with an RHS brace welded to one side of the chord and supported by a flat plate or another profile welded to the opposite side of the chord, the lower effective h_1 on either side and $\chi_{0.5}$ should be used for the determination of the resistance. (3) For X joints (class b) with an RHS brace welded to one side of the chord and supported by a flat plate or another profile unwelded to the opposite side of the chord, the lower effective h_1 on either side and $\chi_{0.7}$ should be used for the determination of the resistance. The $\chi_{0.7}$ value could be obtained from: $\chi_{0.7} = 1.12 - 0.017 \frac{h_0}{t_0} \sqrt{\frac{f_{y0}}{355}}$ Here, the chord cross-section slenderness should not exceed class 3. (4) For other unspecified X joints (class c), the resistance should be determined using $\lambda_{1.0}=2\lambda_{0.5}$, see Table 21. 		
The cross-section slenderness of class 1, 2 and 3 is defined in national standards.		